

JANUARY 2, 1956

50 CENTS

AVIATION WEEK

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Inertial Guidance:
Out of Laboratory
Into Missiles—Part I

•
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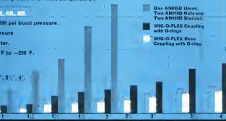
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WEIGHT COMPARISON CHART
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WIG-O-FLEX COUPLING, WIG-O-FLEX HOSE

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RESULTS: Successful



PHOTO BY BOEING



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Sooner

The Boeing KC-135 jet tanker-transport is of vital importance to the national defense. The prototype of this airplane—America's first jet transport—is shown above. To produce and deliver KC-135s to the Air Force as rapidly as possible, Boeing enlisted the aid of Twin Coach Aircraft Division as a subcontractor for major sub-assembly assemblies.

This important assignment typifies the way many contractors rely on Twin Coach Aircraft Division. For Twin is staffed with experienced aircraft specialists in design and build tooling . . . with experienced aircraft production personnel . . . with experienced aircraft supervision and management.

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64-70



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MISSILE SYSTEMS AERODYNAMICS

The rapid growth of missile systems technology has placed new demands on the ability of aerodynamics scientists. At Lockheed Missile Systems Division, new advances are required constantly in the aerodynamic analysis, aerodynamic design analysis, flutter, aero-elasticity and flight dynamics.

Continuing experimentation at Jet Propulsion Laboratory, Ames Aeronautical Laboratory, Gas Dynamics Facility at Princeton University, Naval Supersonic Research Laboratory at Massachusetts Institute of Technology, Langley Memorial Laboratory and other centers supports this program of expanding aerodynamic activities.

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- **Experimental Aerodynamicist** to assist in planning and reporting on experimental programs. This position requires one to two years' experience in wind tunnel testing or full-scale and free flight model testing.

Those who wish to advance their professional stature, while contributing to a group effort of utmost importance, are invited to write.

Lockheed MISSILE SYSTEMS DIVISION
research and engineering staff

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ASST. MANAGING EDITOR (Production) David S. Anderson
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We are now entering the most critical phase in the military research and development race we have known since the post decade against the Communist powers. Our monopoly on atomic fission and fusion weapons is gone and our monopoly on effective aerial delivery systems is ending.

In this period of atomic deadlock, when both we and the Communist powers will have the capability of delivering a devastating atomic attack, the need for dynamic drive through scientific frontiers to develop new and unobtainable superior weapons will become paramount both for enforcing peace in a troubled world and avoiding defeat in an atomic global war. Our survival in both cases and we will depend largely on the speed and skill with which we run this research and development race against our international competitors.

Delay is Fatal

In formulating and pushing a wise and effective research and development policy for this critical period we are already late. Any further delay will prove fatal. Many of the scientific and military men most familiar with the tools ahead are driving hard in the right direction. But there are too many disturbing signs that top level policymakers in the Pentagon, White House and Congress do not fully understand the problem. They appear to be approaching it shuffling with obsolete concepts that will make any effective action impossible.

The most dangerous of these concepts is that of the "constant level research and development budget." This is the current policy of the Pentagon and the cooperative branch of the Government and has forced stress first as Congress. This policy is aimed at keeping military research and development expenditures at the same level—about \$1.2 billion total, of which \$400 million is for the Air Force—for the foreseeable future. Actually, budget figures for USAF research and development now lag its goal in more than \$500 million, but this is a bookkeeping change and does not represent any real increase in the program over previous years.

This "constant level" concept is based about the Pentagon with a vengeance that implies it will assure any technical superiority in the weapons development race.

Nothing could be farther from the truth.

Technical Growth Rate

For technical progress—in as most kinds of natural progress made by man—in a geometrical progression. Plotting the natural growth curve of such key indices as the gross national product, population and electrical energy

available for industrial use during the last 50 years will show clearly that they follow the same geometric progression as astronomical progress.

To follow the natural growth potential of this progression, we must achieve more technical progress during the next five years than we have since the beginning of powered flight. That will be possible only if we have available an expanding research and development budget that follows the same natural growth curve instead of the flatline zig-zag of constant research funds during the past three decades.

The technical revolution that hit aviation with the advent of the gas turbine powerplant and rocket weapons is spawning new problems faster than our present research pace can solve them. Without speedier progress, we will actually lose ground by falling behind the pace established by natural growth. Each new scientific achievement in astronomical research creates a new family of problems that require solutions. Transmittal in aviation, the gas turbine in propellers, atomic fission and fusion in aerial weapons, are all typical of basic new scientific achievements that have expanded and intensified our research and development problems.

To maintain a constant level of research expenditures at an artificial, politically-dictated rate on the face of our expanding scientific problems is the sincerest folly. If it becomes an accepted national policy, it will force our scientific-engineer team to run artificial handicaps, as in a three-legged race, against their international competitors. There is no doubt that the Russians clearly recognize the real nature of the problem and are pouring manpower, brains and money into their weapons development at an ever increasing rate.

Decision Now

This is a decision we must face immediately. We have a long history of frantically boosting astronomical research and development funds on a vain effort to meet a crisis that is already upon us. As late as 1957 we were appropriating bigger research and development budgets in a futile hope of meeting the problem of the Korean war when the shooting had already begun.

The research and development investments we make today will inexorably determine the lead of a country we will live tomorrow or whether we will have it at all.

If we fail to maintain a superior pace of scientific development with an adequately expanding research and development budget, we will not only jeopardize the billions to be spent on future aerial weapons systems, but we will also fatally undermine the future of this country as a free nation.

—Robert Hertz

B.F. Goodrich

FIRST IN RUBBER



How to land a bomber in a phone booth

BEFORE B.F. Goodrich engineers designed and built the company's de-ice upper left hand piston, setting a new leader design was a state-of-the-art piston. Laboratory tests of the finished piston-remotely-operated valve system—designed to be made before best design specifications could be confirmed or modified. But now BFG design engineers have a vision. An electronic analog, almost small enough to fit in a telephone booth, translates a landing gear position-sensing data—into a signal BFG engineers convert and deliver back down data to the engine again, avoid "blind alley" runs that could waste weeks or months.

How do you benefit? We give you faster delivery by providing us qualified team of our best. We give you better

landing-ability-in-between-when-quickly-define new engineering and design goals. And you further benefit by actual demonstration tests which verify the engine's electronic remote sensing.

As you sight you see a BFG forged expansion wheel passing a sonic wave over-land test. Service applied by the 500-200 lb. landing machines are being measured by separate sensor parts and recorded on the one circuit.

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WHO'S WHERE

In The Front Office

Ron Adin, C. Dabson, (USN, ret.), an executive assistant to vice president of Aircraft Radio Corp., Roseton, N. J., recently left to be in charge of aircraft communications in Chief of Staff at New York.

John J. Ochs, vice president Contracts Division, American Radio Asia Corp., Roseton, N. Y., left to be in charge of contract management.

Robert E. Morgan, vice president engineering Radio Enterprises, Hollywood, Calif., recently left to be in charge of contract management.

Capt. T. H. Tompkins, (USN, ret.), executive assistant to president of Radio & Radio, Inc., Pasadena, Calif., recently left to be in charge of contract management.

Dr. Felix T. Swartz, Jr., vice president Research, Inc., Cleveland, Ohio, industrial and government research firm.

Honors and Elections

Dr. Edward R. Sharp, president for 1975 of the Institute of the Acoustical Sciences, Dr. Sharp, director Lewis Flight Propulsion Laboratory, National Advisory Committee for Aeronautics, will be installed Jan. 23 at the House, Noid, Utah, New York.

T. C. Langford, Jr., vice president and chief of the product Control Division of National Association of Airframe Engineers, Gardner, selected first vice president and vice president.

Max M. T. (Admiral) Davis was elected president. He served two previous terms of membership. **Willy C. Brown, Jr., C. C. Co., Jack Hughes, Mr. Francis W. Noble, Vernon P. Boring, Randolph, John F. Vintay and Henry G. Wright, Jr.** were members.

Mr. George S. Brown, Jr., was president and vice president. **Carl A. Fritzel, Dr. Leslie A. Ryan, director University of Texas, Institute of Aeronautics, Mr. Henry G. Coffey, George H. Brown, Jr.,** were members. **William C. Chag, Leon E. Esterline,** vice president, was elected to the first of his 100th birthday. **Walter P. MacCadden, vice president, I. G. Langford, Jr., Donald W. Douglas, Jr., Richard S. Boudry, William R. Kupper, Roger W. Kates and A. Paul Vintay.**

Changes

Dr. Shelden B. Miller, manager, Missile Systems Laboratory, Santa Ana, Calif., will be replaced by **William C. Brown, Jr.,** director of International Telephone & Telegraph Corp.

Walter H. Brown, Jr., manager, Missile Systems Laboratory, Santa Ana, Calif., will be replaced by **William C. Brown, Jr.,** director of International Telephone & Telegraph Corp.

Ono A. Dwyer, manager, Corporate will be replaced by **William C. Brown, Jr.,** director of International Telephone & Telegraph Corp.

INDUSTRY OBSERVER

Northrop Aircraft may be getting sizable production contract for Seahawk with possible as high as 50 to 60 aircraft per month.

Proposals for all-weather intercepter version of the Lockheed F-39H and North American F-107 agonistic fighters have been submitted to the Pentagon as possible backup to the General Electric F-102 in the Air Defense Command equipment inventory.

Cessna's F-102B equipped with a Pratt & Whitney J75 turbojet is scheduled to make its first flight by next fall.

General Electric's Special Defense Projects Department, now working on the improved all-weather intercepter version will bring Schenectady, N. Y., to a Philadelphia suburb. The Special Defense Projects Department, formed six months ago, drew upon General Electric's Hercules missile project for technical personnel.

Av. France is interested in development of chief-carrying vehicles that would be first used in its aircraft to conduct many roles.

Republic Aircraft has hired a group of scientists who recently resigned from Lockheed's General Motors Division after a sharp policy disagreement (AW Dec. 29, p. 16). They will work in Republic's Advanced Development Division which has the General Motors Division.

Rand Development Corp. has developed a technique for joining titanium with nickel or cobalt. The technique is General Electric's Tool Co. Fixing is necessary because titanium parts fail early under friction and may fail. The Rand process has been licensed over to Cleveland Parasitic for further study.

Military electronic equipment and component deliveries during 1975 amounted to about \$2.4 billion, an increase of \$100 million over 1974.

Radar equipment, built in the U.S. to Russian standards on specifications furnished by USAF, is being evaluated against comparable U.S. radar by at least one electronics manufacturer.

USAF still is expending its wealth of capital from Japan depots to aircraft bases. More than 500 aircraft a month are being shipped by air to the Pacific, Europe and Alaska.

Air Material Command's depot maintenance contract to private industry increased from 19 to 46% of total value in 1975. Industry now reflects approximately \$254 million annually for this work on aircraft and various components.

Recent transmission of data from orbiters, as defense relies to avoid an acute traffic control center in the Boston/Warwick, may could be accomplished with an initial installation cost of approximately \$145,000, and an annual operating cost including maintenance of approximately \$112,000 according to a research completed study by Kodak Technical Operations for Aeronautics. The transmission circuitry is non-critical due to "store-and-forward" receiving techniques, such as the system developed by Muller, Raytheon & Deere, Inc.

U. S. Army will move its field artillery battalions equipped with the Chaparral guided missile to Europe in the first four months of 1976. They were based at Ft. Bliss, Tex.

A turbojet, in the compression section of the General Electric J75 engine has caused the Air Force to ground the F-39H Seahawk. Less than a hundred J75 engines have the turbojet, and most of these are spare parts. Only minor modifications are required to fix the compressor, USAF said.



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Washington Roundup

More CAA, CAB Fees?

Question of additional fees for Civil Aeronautics Administration and Civil Aeronautics Board services to industry will come up in the new Congress—but whether it will be done about it. The staff of the Senate Government Operations Committee has prepared a report as the first public of all government agencies. But the committee takes the position that it is up to the committee with specific jurisdiction—the Commerce Committee for Commercial Aeronautics—in legislation. Commerce Committee had also heard the case to the Government Operations Committee.

CAB Candidates

T. B. Wilson, former chairman of the board of Trans World Airlines and now Deputy Under Secretary of Commerce, is understood to be the choice of Louis Rifeffeld, Under Secretary of Commerce for Transportation, to replace Ben Bates as chairman of the Civil Aeronautics Board. It is generally accepted that Bates will leave the Board this month for a federal judgeship. Sen. Warren Magnuson (D-Wash.), chairman of the Senate Commerce Committee which will consider the new appointee, is understood to look favorably on the appointment of Wilson, who served as vice president of Alaska Steamship Co. from 1915 to 1935. Sen. Mike Mansfield (D-Mt.) chairman of the Commerce Aeronautics Subcommittee, is reportedly supporting the appointment of Richard Lloyd Jones, publisher of the Tulsa Tribune, to Bates's place.

Small Business Gains

Watch for relief arguments from USAR in public small business has nothing to complain about, when Congress starts its annual jobbing into law contracts are let and who gets them. In last four months of Fiscal 1956 legal procurement from industry was down to \$483 million from \$1.1 billion in same period of 1955. Yet the contract trends to small business jumped from \$117 million to \$142.4 million. For Fiscal 1955, small business was awarded \$777 million in contracts, up \$20 million from 1954.

Higher CAB Salaries

On the agenda for prompt Senate action is legislation raising the salaries of Civil Aeronautics Board members from \$14,800 to \$20,000 a year, and the CAB chairman to \$20,000. The Secretary of Commerce would be raised from \$22,900 to \$25,900 a year, and the Under Secretary for Transportation from \$17,500 to \$21,900.

The salaries of the secretaries of the Air Force, Navy and Army would be increased from the present \$18,900 to \$22,000.

CAA Telecommunications Transfer

Civil Aeronautics Administrator Charles L. Jones has made a decision on the proposal to turn over CAA's telecommunications system to private interests. Rep. Robert Milligan (D-W. Va.) claimed that it was not "reasonable to conclude" that the issue of 1957 B. Lee in CAA Administrator was detained by Lee's

refusal to approve the transfer of the communications net (AV Dec. 20, 1955, p. 11).

The actual proposal for transferring the communications net to private interests was among the recommendations made by the subcommittee on Capitol Hill. McClellan and Faget in its survey of CAA operations. The report, which cost \$104,000, was ordered by former Under Secretary of Commerce Robert Murray. The report has never been publicly released and has been a source of controversy in the CAA.

Plans of the Administration generally has been to buy over or lease, Government operations as possible to private industry to eliminate duplication. Lee claimed the transfer would be illegal.

Nonsked Fight

The scheduled airlines will take their case against Civil Aeronautics Board's decision opening the air transport fee field to make participation by non-scheduled airlines in Congress.

Their position will be that the Board decision is "in disregard of the mandate of the 1938 Civil Aeronautics Act."

The carriers' critics noted the authority granted the six supplemental Air Carriers to operate on scheduled flights a month before and two years. The scheduled airlines noted strong objections with CAB on the decision and the fight will go on in the courts as well as Capitol Hill.

Army Information Change

Veterans Army public information work is disrupted by new staff setup which places their activities under the Deputy Chief of Staff for Personnel. While this step is not so immediate trouble because many have been transferred to other units, it will still affect officers, their activities will be handicapped. Top personnel post could be filled in later by an officer with no credits in press relations. He could handle PIO matters in that contact with other Army branches, effectively cut off sources on Personnel staff, an important source of Army news. Army at the week of May 1956 Gen. S. Meloy, Jr., to replace Brig. Gen. Theodore S. Ryan as Army Chief of Information was once set off new effort to have his office responsible directly to the Chief of Staff.

Indian Bilateral

The U. S.-Indian bilateral agreement dispute may be settled soon. A U. S. delegation returning in complete agreement is expected in New Delhi early this week. After extensive consultations held in India last fall, the U. S. delegation brought Indian proposals back to Washington for study and has since returned to review the proposals.

Indian sources expect an agreement to be signed which represents a compromise between the American principle of letting economic factors determine satellite frequencies and the Indian desire to increase its share of the total market by controlling frequencies. Settlement of differences is a compromise between these two represents a cold and a long period of talks negotiations between the two governments.

—Washington Staff

USAF Aligns Industry for Atomic War

Industrial production readiness policy outlines concepts to compress, accelerate weapons output

By Claude Witte

Washington—The Air Force gave the U. S. overall industry a deadline but seeks to achieve production readiness in case of atomic war.

USAF Secretary Donald A. Quarles said, in effect, that the threat of attack in Soviet long-range strategic bombardment could require the industry to be on a constant state of readiness. USAF's top priority weapons systems must be "ready" and "available."

In case of war, every available piece of hardware must be thrown into the breach at once with special emphasis on weapons needed by the Strategic Air Command and Air Defense Command.

USAF spokesmen said there are no general Production Readiness Policy has been approved by the Department of Defense and that it is complementary to all war plans approved by the Joint Chiefs of Staff and the National Security Council. They said it does not conflict with any of the plans and is designed to carry them out.

Essential to Air Force

There was no open indication of when the Atomic Energy Act will take similar steps or why the new readiness policy was not promulgated by the Defense Department for all branches of the service. Some Pentagon observers

contended that preparations for a 30-90 day war are necessary under present plans adopted over protests of the Army and Navy.

USAF, on the other hand, pointed out that the new readiness policy is most essential for its mission, largely because SAC and ADC will be the first to act in case of a general war. Far from being a hindrance, the industry needs in case of a long-term war, USAF said, the new plan will speed mobilization in this case as well. This includes the Army's need for aircraft purchased through USAF channels.

Two Capabilities

However, USAF emphasized, the new approach is essential "because, for the first time, we have broken, and are breaking, the link between potential threat of devastating destruction at the outset of general war and, as an emergency, we are to survive and measure the devastation to our nation, the traditional concept of a prolonged central buildup after attack must be replaced with a readiness program."

For the aircraft industry, the new policy calls for perfection of two new capabilities:

The first is a Production Compression Capability (PCC). This will assure maximum all-out production of strategic and defense weapons in the first 90 to 180 days of a general war. The

program must go into effect instantly in case of an emergency.

The second is increased production to be achieved "whenever the need arises," with working plans for a minimum week work and during the normal inventory of finished components, equipment and sub-assemblies on hand on the available plant. It added that potential output can be increased by having a larger inventory of finished items on hand and that this will be done in plants where it is possible on the basis of personnel, cost and criticality of the weapon items.

Here are the assumed wartime set-up steps in a plant ordered to set up a PCC program:

- The assembly plant will remain intact.
- No additional personnel will be available during the all-out production period.
- There will be no support from Government financial or technical equipment (GFAE) production, subcontractors or vendors during the period.
- Loss of all except local transportation facilities.
- Communications facilities limited to high-priority messages.

Second type of war readiness effort provided for is the new readiness policy is Production Acceleration Capability (PAC). It will provide for better utilization of existing or supporting resources in a predetermined peak rate in the shortest possible time.

This can be accomplished, Quarles said, by minimizing stockpiles, expediting production capabilities and increasing the inventory of finished and sub-fabricated materials at prime, associated and subcontractor facilities. This capability must be trained in advance of an emergency, that is, it can go into effect immediately.

- The increased inventory of materials will be purchased with deferred funds. It will be ordered as the last requirement of assembly is produced under the current program.
- Production accelerations will be applied on a case-by-case basis dependent upon production status, duration of material production contracts, additional stocks and availability of the weapon item or supporting prime contract.
- Production accelerations plans will be of greatest value in limited wars or geographic expansion.

Category Plans

The new USAF policy emphasizes that resources, which exist in excess as well as production facilities, are limited. For that reason, the program will be applied chiefly to the aircraft indus-

Industrial Readiness

try, but on the basis of selective plan-

ing. Major aircraft and missiles will be divided into categories, grouping each in severity during hostilities. The weapons to be classified will be those on the active Missions Wartime Aircraft and Guided Missile Requirements List, forming a new Strategic Planning List issued in May and November of each year.

USAF spokesmen said prime contractors will be responsible for participation of their subcontractors in the program.

- That it is not anticipated that these will be changes in existing contracts.
- Selected systems in Categories A and B will have contractual provisions in future procurement.
- For top-priority items in Categories A and B, specific steps to achieve readiness will include establishment of all links in production chain of a general order war.
- Improve industry's ability to maintain or rapidly return production of critical aerial and propulsion weapon systems and related support components in case of general war involving severe industrial damage.
- Monitor the health of the aircraft industry as necessary to fulfill the needs of the Air Force.

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Renegotiation Act and Profits:

Analysis Minimizes Net Worth Factor

Washington—An analysis of the net worth factor in renegotiation could cast doubt on taking away as excessive profit everything above a supposed reasonable rate as net worth would be a sharp departure from an earlier position and a disregard of the intent of Congress.

The analysis was prepared by John T. Koehler, who served as chairman of the Renegotiation Board from 1951 to 1953, and Philip Nichols, Jr., general counsel of the board from 1951 to 1954.

Both are now practicing law in Washington.

The aircraft industry in particular has been jarred by actions of the Board which disassembles a new and tougher policy on excess profits. The Board has not publicly released its reasons for its actions and reasons in recent renegotiations, causing speculation that net worth may now be given more weight than formerly.

For that reason, the program will be applied chiefly to the aircraft indus-

try in order to help in the effort. USAF will maintain close ties with the industry to make sure all plans are realistic and feasible.

While it is not anticipated that these will be changes in existing contracts, selected systems in Categories A and B will have contractual provisions in future procurement.

- For top-priority items in Categories A and B, specific steps to achieve readiness will include establishment of all links in production chain of a general order war.
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USAF spokesmen emphasized that weapons placed under the PAC program must be beyond the stage where excess profit engineering changes are expected. In the case of a weapon under PAC, the

contractor will be referred to make all source payments of war damage and similar calculated parts but there will be limits placed on the stacking of finished parts and assemblies. Raw loss packages may go up to \$175.

In the case of PCC items, it is assumed that subcontractors will be kept and there will be severe industrial damage.

Natural inventory will be maintained with the idea that the manufacturer will be able to produce the maximum return from what is on hand. He will be expected to offset total aircraft, expeditious delivery of weapons on the line and do anything else possible to get airplanes into operation.

It is not anticipated that the readiness program will add materially to USAF costs. In fact, spokesmen say, use of the nation's stockpile of jet engines and jet replacement engines and much more.

Another major USAF aim is to get new lines into full production, get the most efficient lines in operation to keep the "hot line" as its most useful concept.

Secretary Quarles ordered an immediate review of all current and emergency production of weapons and supporting systems. He said adjustment to the new policy "is to be accomplished as rapidly as possible but with due regard to economic considerations and the various adverse effects of sudden change."

ever had the language of the House Bill become law, there could have been a policy available to attempt to determine the reasonableness of return on the contractor's net worth in even cases, and to get it more widely distributed in World War II.

Renegotiation Act History

Koehler and Nichols traced the history of the 1951 act which sets general renegotiation objectives.

The act was drafted and introduced with a capital and net worth factor reading the same as that in the 1945 act. No doubt it was the influence of the Brewster (Thomas Cox) House Committee which led the Ways and Means Committee of the House to advise its amendment of the latter to read as follows: "Reasonableness of return on net worth with regard to the amount and source of public and private capital employed."

Let the Senate Finance Committee disagree with the House Bill. The committee took the words "reasonableness of return on" going as the factor as we have it today. The committee opinion said that if amended the House Bill said too restrictive since the reasonableness of return on net worth might not in too many cases be an adequate measure of its profits.

The Finance Committee bill was carried over into the 80th Congress but to the House that was a general mandate to change prior practice concerning capital and net worth.

Both houses noted that contractors

Aircraft to Receive Priority Ratings

Under an Industrial Production Readiness Policy, Air Force will assign priority ratings to meet aircraft and guided missile needs. Ratings will be determined on this basis:

• Category A. This category will include the most important Air Force war and defense aircraft and missiles which are essential to the national defense. Selection will be based upon recommendations of the Air Staff and Air Council, after the following factors have been considered:

- Criticality to the national or regional security.
- Performance capability and versatility of the system.
- Current production status and general probability of the system.
- Items vital to the industrial structure required to produce the system.
- Current inventory quantity of the system.
- Category B. This category will include weapons systems and supporting systems included in the Missions Wartime Aircraft and Guided Missile Requirements List, which have not been designated Category A. Under existing military emergency conditions, servicing plans for the category would be appropriate for support production of Category A systems if required. Category B systems will be kept in production under normal conditions.
- Category C. This category includes all remaining weapons, aircraft and supporting systems for which production planning is desirable. Category C systems may not necessarily be kept in production in case of general war involving severe industrial damage. However, capacity for this category could be of great value in rapidly accelerating production during a limited war, emergency mobilization phase of general war, and to provide systems that can be reprogrammed to support production of the A and B categories in a general war involving severe industrial damage.



Chessie Takes to Air

Keeping pace with modern reactor transportation, the Chesapeake and Ohio Railway has fitted Chessie with DC-3's but used less Capital Airlines. "CRA's one plane will make one office to cover more efficiently one 3,000 miles of railroad operations," said Walter J. Touhy, the company's president. Touhy (right) is shown at delivery ceremony with Capital's President James H. Cheswood.

must be given to the rate of return on net worth. This is to afford a proper incentive for the investment of equity capital. But the contractor must be afforded enough return on his debt capital, too, and, to afford the contractor incentive for raising rate debt and retiring the less of equity capital.

"There will be cases when the rate of return on net worth will be so high that it will give rates for money, but in most of these cases that return will have to increase to a high level to avoid doing violence to other factors which are of paramount importance," Kockler and Nichols noted.

Both negotiation experts feel that the board should feel free to disregard the return on net worth altogether in these cases:

- When it believes that the book return is misleading as to the contractor's actual cost.
- When it cannot make a reasonable allocation of net worth between non-eligible business and other business.
- When the capital and net worth employed is so extensive that it will afford a measure of the contractor's capacity to contribute to the defense effort.
- The Reconstruction Act requires that the board, in determining excessive profit, must give favorable recognition to the efficiency of the contractor and take into consideration these five factors:
 - Reasonableness of costs and profits.
 - Net worth.

areas of return on net worth as a measure of profits.

- They do not provide equitable stand-ards for comparison.
- They do not take account of replacement costs.
- They do not measure contribution to the defense effort.
- They disregard the real factor.
- They do not treat all capital funds equally.

New Lockheed Facility For Nuclear Research

A large laboratory and two atomic reactors will be the key installations in a new facility Lockheed Aircraft Corp. will build to conduct nuclear research related to atomic-powered aircraft.

The new reactor center will be located at Marietta, Ga. (AW, Dec 18, 1955, p. 9) on a 40-acre wooded site 18 miles away with elevations ranging from sea level to 2,800 ft.

Referenced sources say the laboratory and reactor will be widely spaced and be serviced by a narrow gauge railroad which will be controlled remotely. The entire outfit for the new facility will cost between \$10 million and \$20 million.

Lockheed would make no comment on the subject, but remark at the facility project already associated under a preliminary design study contract on nuclear-powered aircraft with the Air Force. Lockheed first announced the contract in October, 1953.

In addition to studies of nuclear power patterns of structures and equipment subjected to atomic exposure, scientists at the new facility reportedly will investigate factors relating to pilot and ground crew protection and how such protection factors will be handled. They also will consider problems inherent in the production of plutonium as the result of a power crash.

Construction for the design of the new facility has been awarded to Lockheed's Georgia Division to Vitro Corp., which is well established in the nuclear and power-generating fields. Work on the plant is under way. It is expected that full design of the new facility will take from eight months to a year.

At Lockheed's California Division at Burbank, a unit of physicists and engineers are at work to provide Vitro with standards and technical assistance on problems principally relating to shielding and methods of handling materials. In effect, these technicians are helping Lockheed understand the unit as the "house" being built for them.

The physicist and engineers will be transferred to Lockheed's Georgia Division somewhere before the completion of the facilities.

Aeroflot Offers Promptness But No Frills

Traveler reports: Soviet pilots are competent, but passengers must fly without safety belts.

By Harry Schwartz

In Moscow one day last fall, a very plump American correspondent (this writer) got ready for his first ride on a Soviet airplane. As he pulled his legs to stand itself upright, on the nose stem of a few weeks earlier that told of a radio group of Norwegian airmen killed on a Soviet plane crash. He recalled the stories he had read and heard about Soviet pilots who took off without wearing up their engines and who never flew higher than 500 feet above the ground, if that much. The American said he had heard of even less so eager to get into a business that required him to face the prospect of a 500-mile flight in a plane being a Aeroflot, the Soviet airline.

A month and two weeks later, 300 air miles to Aeroflot plane later, the writer met an even more newly arrived American in Moscow. When that reporter began expressing trepidation about the wisdom of flying Aeroflot, the writer was asked to get him back straight away. "Don't be silly. You'll be put to rest as you'll be an American Aviator or KLM Aeroflot, that's a real good saying, even if it does cost all the money."

Mr. comment, if it can be called such, was the product of a couple set of facts. Even one of the half dozen Aeroflot planes I took started in flight on time and landed such at its destination on time as well as in minutes of scheduled arrival time.

Soviet Pilot a Sensible Man

So far as I could judge from my passenger conference, Soviet pilots were all sensible men who arrived up their engines before taking off, who flew between 6,000 and 10,000 feet above sea level and who planned their flight paths as to be in a clear and though sometimes rather often directly over populated areas.

In one way, the Soviet flight crew seemed to know what they were doing and to be just as anxious to land safely as I was.

So far as I could judge, simplicity and common sense were the keywords of Aeroflot's planning and service, subconsciously well as safety. None of the planes I took had safety belts, but they all had stewards and a crew chief. I got something to eat or drink. All the planes I took were essentially the same—two engine aircraft carrying a maximum of 21 passengers in seven rows of seats,



PASSENGERS carrying their own luggage disembark from Aeroflot Ilushin II at Moscow's Vnukovo Airport. Traveler reports that in smaller cities Aeroflot pilots land on dirt strips. Soviet air travel is a simple matter for foreigners, but it is a different story for the Russian who are faced with the poverty system and a lack of ground transport.

Exclusive Report

This article on the stress and competence of Russia's state-owned airline was written especially for Aviation Week by Harry Schwartz, published as Soviet Airlines for the New York Times, who is credit attached from a five-week tour of the Soviet Union. He traveled more than 5,000 air miles over Aeroflot routes.

order-of-air-distribution. Most of the passengers prefer to go sitting, however, or to stretch themselves on the cabin benches across the formation of the plane's vertical position throughout the journey.

For the American or other foreign visitor to the Soviet Union, landing on Aeroflot is a task, not a pleasure. The Soviet travel bureau, attached to all the details. It buys the tickets and delivers those to the passenger. It provides a pre-flight information, to take the foreigner and his baggage to the airport, as well as a special airport waiting room where he can wait comfortably. When the plane lands, there's an instant man and automatic waiting to whisk the foreigner off to his hotel.

begin and all. It's quite simple.

For the Soviet Union, it isn't quite that simple. Soviet citizens bring on Aeroflot planes as a matter of course, people, leaving their engines and leaving to carry parcels. Most of them, as far as I could learn from our conversations on planes, were traveling on "komandirovka," that is on business for some government agency or other, unless they are American tourists to take their own private cars, that seem to have problems getting those visas and their baggage to and from airports. There were no Civil Reserve Air Corps or anything resembling those—visible at any of the airports where I stopped. There were in fact some of the best, however. For people traveling privately, Aeroflot services, but at least a small office in the downtown section of each city and people come there to buy tickets.

No Show! Soviet Style

Regrettably, I should add, isn't necessarily a matter of the Russian passport. The least in 10 kilograms (22 lb.), and I never saw a Russian with anything like that amount as passing an event baggage fee. Women passengers usually carry big handbags and the men big sports bags, but large backpacks. For the foreigners, however, the limit is seven 1 lb. per ounce on every flight.

Judging by some of the conversation I observed before some of our planes took off, Aeroflot has the "no show" problem too without having anything like the elaborate "standby" system available to traveling tourists. The "no show" problem must be particularly severe because Aeroflot's passenger capacity is strictly regulated compared to the size of the country and of its population. At a given, I would judge that Moscow—the air route to Kiev—but less air traffic than that from Cleveland or Detroit in the same time. Aeroflot does such pretty much all over the Soviet Union—at least at major cities. But they kept 50 and would like flying, only main lines operate and even such major cities as Kiev and Odessa, Yerevan and Baku are without direct plane connections. The Russian under, after all, is run Soviet, and Aeroflot apparently prefers to concentrate its main on operating a few major routes in which is possible during the long winter season.

The danger of these sorts during seasons in winter creates a problem of deterioration, of course. This is particularly true because Aeroflot's rates are competitive with those of Soviet "air cut"—the Russian equivalent of Pullman but not quite as comfortable. The whole Aeroflot plane is over five or six times the speed of the 70 with a four

'engine' which runs on Soviet rail-turbine "Tuzh" of which Soviet rail-turbines have been, even in long distance runs—runs to average 15 miles an hour. Just how they solve the distribution problem I can't say, except that there are great groups of people being exchanged. One sign I saw at the Minsk Airport listed those top priority groups of applicants for plane tickets: members of the Communist Party Central Committee, members of the Supreme Soviet (Russia's legislature), and MVD (secret police) field reconnaissance units. There may well be a black market in plane tickets there as it is everywhere else that's so rare.

Soviet Airports

In American standards, Soviet airports are small and have rather limited facilities for passengers. They're working like the most of shops and services available at Lilloret or Lufthansa for the passenger who takes off from, or arrives at, Moscow's Vnukovo airport. Windows at least, but general reasons. Airports at smaller cities don't have such facilities, but Soviet pilots seem to be quite experienced at landing their planes on ground in American pilot would consider a comparison of some American airports. I think they participated in the air accident another than those I've made in some DC to landing at night, American airports.

To sum up, if what you demand from an airline isn't just to get you from one place to another reliably but as compared with competitive service of transport, and with a high safety factor, then Aeroflot would seem to fit the bill. But it is that the quality of his planes, rising at 20,000 feet, pretty slow stream serving small airports and "stoppage" points, this long way. Aeroflot considers that things "happen" but it isn't having any, at least not in his.

Aircraft Employment Reaches 755,800

Employment in the aircraft manufacturing industry in October reached 755,800 of which 514,969 were production workers according to the Bureau of Labor Statistics.

The Bureau reported an increase in employees during October for all companies in the aircraft industry—493,500 workers with average hourly pay of \$14.640, annual contract and gain of 1,000, aircraft production and parts, and 105,000, other aircraft equipment.

The average work week also increased during the month to 41.4. Longer hours and a 2-cent pay on hourly pay brought the average weekly pay of aircraft production to \$611.49, almost 56 more than in October 1974.

Democrats Ready Aviation Investigations

By Katherine Johnson

Washington—The Democratic-controlled Congress has started the Defense and Commerce Departments has listed studies in this Presidential election year, and the aviation industry will be in the midst of the battle.

The day after Congress opens on Jan. 4—Sen. Mike Mansfield (D-Ga.), chairman of the Committee on Commerce, Subcommittees, has scheduled hearings on the firing of Fred B. Lee as Civil Aeronautics Administrator. Mansfield has already made it clear that the hearings will go to beyond Lee's dismissal into charges that this was only one more in a general plan of United States Secretary of Commerce Louis Rothchild to control not only CAA but also the Civil Aeronautics Board. Mansfield will also challenge the appointment of Joseph Mammie on two grounds that he is dominated by Rothchild and that he is not specifically a Democrat. Under the 1978 Civil Aeronautics Act the Board position normally held by John Lee must be filled by a Democrat.

Plan of Attack

The congressional attack on the Defense Department will take two paths:

- Failure to adequately increase military strength in order to make budget savings so that the Administration could prepare a balanced budget.
- Failure to make adequate steps in the defense budget by expanding the military procurement program, of big business Rep. George Minton (D-Tex.), Chairman of the Military Appropriations Subcommittee expressed the concern of the Democratic Congress that they a transfer both in Congress and use.

"The profits of aircraft manufacturers have been soaring in though they were not expected."

"This has in some instances reached astounding levels. The industry is bringing back a much less... The Securities of Defense in a gross in the field of business and management, but it is far too locked up my heads together an industry in an effort to get defense cuts down to profitable ends through better contracting. I have not heard about it."

The Democratic attack on military plant contracts will be generally limited to the long business management of Defense Department, until Secretary of Defense Charles Wilson, but defense contracts, generally aircraft contracts, will also be examined. Here is the investigation schedule:

- House Armed Services Investigating Subcommittee will open public hearings on in January, calling in such aircraft manufacturer, aircraft engine manufacturer and major aircraft sub-contractors to determine the results of Defense Department procurement policies on the profits and valuations of the companies. The expansion of the industry, necessary to achieve a 130% Air Force, has resulted in aircraft and aircraft.

- The subcommittee is headed by Rep. Edward Heise (D-La.). It is being headed by Rep. Carl Vinson (D-Ga.), chairman of the full committee and the congressional overseer on profits in aircraft contracts, reportedly with the full approval of House Speaker Carl Albert (D-Tex.).

The subcommittee was that it is in fact as it is present in the public air contract. Secretary Wilson's recent statements that the new money

Department procurement policies

- The House Appropriations Subcommittee, which will begin hearings on the Fiscal 1977 defense budget in closed sessions in February, has already reviewed one of its primary issues of questioning the percentage profit of aircraft manufacturers based on net worth.

- The Defense Department, as a result of a congressional directive, is due to publish a listing of the 100 biggest defense contractors by the present fiscal issues through June of 1975, and from July, 1975 (the start of the fiscal year) periods through June 1977. Investigations into the concentration of defense business on the grounds of the Senate Preparedness Investigating Subcommittee headed by Sen. Ladd (D-Tex.), the Senate counterpart of the House Subcommittee of the House. The Small Business Committee of both houses also is closely following the concentration of defense business.

- Senate Small Business Committee will open hearings on Thursday on its findings that Defense Department tend toward repeated contracts and may favor competitive bid contracts.

Commerce View

- Sen. Stuart Symington (D-Mo.), former Secretary of the Air Force, has taken, and undoubtedly will continue to take, the lead in Democratic criticism of the Administration for raising military strength—Munroe and Army growth factors with deficit support as well as USAF strength—a price to budget constraints.

Democrats, disappointed with the Secretary Wilson's recent statements that the new money

Military Aviation Funds	OBLIGATIONS (000 Dollars)		EXPENDITURES (000 Dollars)	
	Unobligated	Obligated	Unexpended	Expended
	July 1, Oct. 31	Balance, Nov. 1	July 1, Oct. 31	Balance, Nov. 1
Acash, Excess, Post				
Air Force	-103,000	9,181,771	1,750,310	13,829,566
Navy	1,400,000	2,045,179	540,391	3,476,086
Air	3,000	300,100	467,179	1,000,000
MDAP	-6,188	-	10,648	479,363
TOTAL	-104,300	12,527,050	2,309,428	19,983,114
Capital Matters				
Air Force	34,770	1,072,156	125,550	1,425,829
Navy	95,395	330,300	51,899	357,074
Air	108,114	330,036	83,815	919,591
TOTAL	238,279	1,732,492	261,264	2,642,794
Defense & Communication Research				
Air Force	38,215	850,138	586,155	1,450,135
Navy	1,000	1,000	1,000	179,130
Air	7,086	189,257	20,339	325,228
MDAP	11,010	-	21,351	-7,969
TOTAL	64,311	1,139,395	634,645	2,374,544



New "Anchor" for F-84F

New design products for F-84F Thunderbolt fighter bomber type both with a force of 2,000 lb. and helps stop the aircraft in 2,000 lb. 1,000 lb. less than the type equipped in normal landing without chute. The products is 56 ft in diameter and has 38 ft behind plane.



Beech Tests Jet Version of Mentor Trainer

Four Beech-built jet-to-the Model 73 two-seat tandem, turbo-propeller-powered and built by the Wichita, Kan. firm is a private venture, but started its flight tests. Based on the Model 65 Mentor (1,540), the Model 73 is powered by a Continental J69-10 at approximately 300-hp thrust. It has a top speed of 251 knots at 11,000 ft, a stall speed of 60 knots and weighs only 4,321 lb. The Model 73 is designed to provide students with familiarization in aerobics, night flying, basic instrument flying or addition to dual instruction.

request for Fiscal 1957 will exceed Fiscal 1956 appropriations by \$2 billion and that planned expenditures in Fiscal 1957 will exceed those of Fiscal 1956, says the statistician this way.

- **Expenditures.** The Administration is holding these down in Fiscal 1956, which ends June 30, 1956, for the express purpose of paying the way for a balanced budget and his activities as the director year.
- **Appropriations.** With a big appropriation for Fiscal 1957, available July 1, 1956, the Administration will have a large fund available for commitments in July because during the crucial closing months of the 1956 presidential election.

On the matter of Defense Department building class spending during Fiscal 1956, Democrats have that to show: During the first four months construction of aircraft contracts exceeded one contract by \$244 million. The development program despite a report approved by House and Senate clerks that although not expenditures for aircraft would be "accelerated in the event possible consistent with present technological developments and efficient application."

Three key senators on civil aviation face a election this year, so far without opposition. They are:

- **Sen. Warren Magnuson** (D-Wash.), chairman of the Commerce Committee.
- Gov. Arthur Langley** and **Under Secretary Connerly**; Walter Williams have been mentioned as possible Republican opponents, but neither has indicated any intention. Magnuson defeated Williams by a wide margin in the 1950 election.
- **Sen. Mike Mansfield** (D-Mont.), chairman of the Commerce Aviation Subcommittee.
- **Sen. George Southern** (D-Fla.),

chairman of the Special Commerce Subcommittee on International Aviation. Former Sen. Clark Pepper, defeated by Southern in 1950, but said that he will run for a Senate seat in either 1955 or 1958.

After the opening attack, led by Norcross and directed at Rothchild, the Commerce Aviation Subcommittee is scheduled to take up the following matters:

- **An Independent Civil Aeronautics Administration.** Members are assumed that he will sponsor legislation separating CAA from Commerce. The industry generally, including scheduled and non-scheduled airlines, are expected to take the position that separating the agencies now would result in a more hazy than ever. Air Line Pilot Association, though, will support the independent CAA plan.

• **New Washington National Airport.** On that issue, Rothchild and Magnuson are headed for another outcome of six years. Rothchild wants to set up a state authority to operate a second airport for Washington. Magnuson takes the position that the delay involved in this would result in their sought inadequate facilities to meet the needs of a low year hence.

- **Less presidential authority.** Legislation limiting the President's authority to veto Civil Aeronautics Board decisions to the foreign relations and national defense aspects of international route case approach will have clear sailing through Congress.

The White House is understood to favor the measure. This would eliminate White House jurisdiction over routes to territories completely.

- **Permanent Alaska routes.** Magnuson will push legislation to give the two circuits—Pacific Northwest Airline and Alaska Airlines—permanent certificates.

This will probably open consideration of permanent certificates for helicopter services. The committee took the position last year that these should be a further trial period for helicopter services before permanent certificates.

- **Industry representation on international bilateral airline negotiations.** The special representative headed by Southern will hold hearings early in the session on the German bilateral agreement. A report calling for industry representation on international negotiations is expected. The representation already has the support of CAB Chairman Ross Rulley.

Air Force to Study Radiation Effects

A new atomic radiation test cell, which will be used to gain information about the behavior of aircraft materials and equipment under atomic and gamma ray bombardment, will begin operations within the next year at the Wright Air Development Center's test tank laboratory. A contract for 75,000 cubic feet of gamma radiation flux is the largest single amount of Cobalt-60 (a neutron) can be applied upon tested objects in the 136,712 ft³ shielded testing space. This could simulate the environment of equipment in the vicinity of the reactor of an atmospheric aircraft.

- **Some of the effects of radiation which will be studied in the 514,300 facilities include:**
 - **Tendency of lubrication oils to change viscosity and composition.**
 - **Deterioration and bubbling of transparent plastics.**
 - **Interactions with the operation of transmitters and gas filled tubes.**
 - **Deterioration of electrical insulation.**
 - **New methods of releasing oxidizers.**



S-14 Makes Second Bid in Jet Trainer Market

A second Beechcraft S-14 turboprop aircraft jet trainer possible and of the Beechcraft Model 73, arrived in the U. S. late last month for a demonstration tour of Air Force and Navy training bases. The first S-14 brought to the U. S. by Fairchild Aircraft, which built the Beechcraft to manufacture and sell the trainer, landed at Pittsburgh, Pa., on Oct. 25. Powered by a Rolls-Royce Dartment "50" turboprop jet low gas turbine engine with 1475 lb. thrust, the S-14 has a top speed of 400 mph.

European Aviation Group Forms, Plans to Increase Air Traffic

First action of the European Civil Aviation Conference ended last month at Strasbourg, France, with agreement on a constitution and future working methods.

The constitution provides for the conference to call its own meetings and fix its own agenda. It will hold annual plenary sessions, the main task of which will be to advise the development of intra-European air transport, in order to promote coordination between national and an orderly development of traffic.

Chairman will be maintained by the Conference with the International Civil Aviation Organization. Financial arrangements for the conference will be considered at a meeting to be held immediately with the 19th session of the ICAO assembly, at Geneva, Switzerland, in June 1956.

Representatives from 15 European countries met at the first session of the conference, which was called by ICAO on Nov. 28, 1955. The conference elected Pierre Vanier of Belgium as president, and Luis de Aranzaga of Spain, Ali Hakan of Norway, and Antonio Antonides of Turk as its three vice presidents.

First order of business was a discussion of the need for standardization of regulations and a strengthening of economic arrangements in Europe. Three minutes were devoted in support of better coordination and liberalization.

- **Frequency of routes in air traffic network in European networks.**
- **Utilization of aircraft in lower than in the U. S. domestic routes.**
- **Revenue-producing ability of European airlines in considered poor.**

A lengthy exchange of views took place in connection with a multi-lateral agreement for intra-European scheduled services and arrangements for an orderly solution of the problems involved in the interchanging of aircraft and helicopter services.

The conference prepared a draft multi-lateral agreement for solution of the problems involved in the expansion of the non-scheduled services in Europe. This agreement would establish freedom of operation for non-scheduled European airlines in the following categories:

- **Aircraft engaged in humanitarian or emergency missions.**
- **Transportation of passengers in air taxi—non-routine service aspects of air passengers.**
- **Charter flights when there is no table of flight.**
- **Any flights which have a maximum frequency of once a month.**
- **All freight and passenger operations between airports which have no reasonably direct connection by scheduled services.**

The delegates were split on how to substitute a multi-lateral agreement in

place of the present bilateral agreements in Europe by exchanging commercial rights of intra-European scheduled air services.

Some delegates expressed that more progress immediate results could be obtained by concentrating on "cooperation between airlines" and that the role of governments should be limited to the facilitation of such cooperation. This position held that, in fact, more rapid progress is being made through bilateral progress in cooperation between airlines already in operation than the last 10 months.

Other delegates expressed the opinion that cooperation "in the best interests of governments and airlines" in the policy which would best obtain the greatest improvement in the situation. This caused a formal declaration by the states that were represented that they favored direct services between states and would refrain from opposing the establishment and operation of air services of other member states unless they considered such services actually harmful to their own national interests or decided that they did not serve the best interest of the states.

The conference also exchanged views on the problems raised by the development of helicopter services and helicopters. It was decided that improvements are needed in air as operating economy, safety, all-weather operation and noise reduction are concerned. The question of helicopter services will be on the agenda of the next ICAO an aviation meeting in the European-Mediterranean region.

Regarding the interchanging of aircraft, the conference proposed an agreement by a "study group."



THE RECENT ARDC-Industry Conference in Baltimore was attended by 400 persons. Among them, left to right: W. J. Fabron, General Corp.; W. H. Wilson, Hercules Electronic Corp.; W. E. Bebout, president, Tascody Inc.; Ross Adams, W. A. Sikorski, Boettler.

ARDC-Industry Conference



LT GEN CLARENCE S. BYRNES, USAF Deputy Chief of Staff, Montreal, and Ed Webb, Boeing Airplane Co. representative, meet during conference talks.



ARDC REPRESENTATIVE Maj. Gen. Floyd B. Wood, Deputy Commander for Research and Development. Others in picture are: L to R, Ted Stenerik, American Machine and Foundry Co.; S. N. McDonald and J. S. McDonald, president of McDonnell Aircraft.

Prop Phasing Cuts Noise in Transports

A new answer to the problems of cabin noise and vibration is being provided by the nation's three major manufacturers of large propeller-driven aircraft: Hamilton Standard Division, Atlanta's Aircraft Products Operations and Curtiss-Wright's Propeller Division.

The new development not only keeps all propellers turning at the same rate (the basis of prop synchronization), but keeps them at a specific angular rate from one to each other. It is called Struckto phase, Phase Synchronization and Struckto Phasing Device, respectively, by the three companies.

Hamilton Standard now Struckto phasing provides two major benefits. It appreciably reduces cabin noise and vibration, and also makes (reduces) noise more acceptable to the human ear. Instrumented flight tests with Struckto-phasing on a 3-engine Constellation and a DC-7 showed cabin noise reduction of as much as 16 decibels, according to Hamilton Standard Co.'s Cowan 500, noise reduction varied from 10 to 14 db. Although the cabin instrument showed a test plane thought the reduction in vibration was more soothing than actual noise reduction, the company says.

Struckto-phasing will be installed on Trans World Airlines' new TWA Super Constellation. The Aeroquip (air) Phase Synchronization system is going into American Airlines' and Eastern Air Lines' new Boeing 747s. Here a new Hamilton Standard's Struckto-phasing works.

One propeller is selected as a master and the other three (in a four-engine airplane) slaved to it.

A magnet is installed on each propeller and a coil in each engine nacelle. When a magnet passes a coil, it sets up a voltage pulse. This impulse travels to an electronic control which compares the time of the impulses from each nacelle. If the timing of the pulses differs from that of the master propeller, the difference is registered and corrected, keeping propellers in phase. The synchronization has a range band of 20 rpm to prevent slippage from stopping in case of master engine failure. Hamilton Standard says the device will maintain phase relationship within 15 deg. Wright has a four-engine installation in 68 ft.

Hamilton Standard says it originally developed Struckto-phasing 10 years ago, logging more than 500 hours on a Navy PB4Y, but defined the idea by case of lack of customer interest. The project has been revived now because of the increased noise and vibration problems stemming from today's more powerful engines and larger propellers.



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X-1 CROSS SECTION (partial top view). 1. Air intake over bulkhead, 2. Cabin air, 3. Forward cabin, 4. Submarine's main tank.

USS X-1, Fairchild's Midget Submarine



THE NAVY'S FIRST midget, with airplane-like deck structure, was built in aircraft factory, left, and tested in Long Beach Sound.



X-1, shown here in first submergence test, can be disassembled for air transport. Primary use will be to test turbo-jet engines.



ORIGINAL TIGER, then designated F11F, was designed according to the new rules and was apparently in level flight.

New Design Changes on Navy's Supersonic

Latest dated design changes in the F11F Tiger reflect Grossman's increasing cooperation with the experimental fighter.

Most obvious present alteration is in the geometry of the vertical tail, which has now a reduced taper ratio and overhangs the tailpipe exhaust.

Wing fences, which originally extended from leading edge to just ahead of the quarter lateral control inboard, now start well behind the leading edge, almost on the extended trailing edge of the deep-root flap. Skins of the fences now is tapered.

Third airplane off the line (No. 135066) has been assigned to Great

man's Experimental Dept. for developmental work. Current configuration of the plane, which is also the latest production layout, is shown on the work's cover.

That aircraft also has been fitted with and without wing fences; the four 23-in. canards and the new vertical tail.

Earlier changes made to the Tiger's six-rated sharp intake:

- **Lengthened nose**, for increased low speed range and reduced drag.
- **Altered canopy**, replacing the experimental one which had a metal structure.
- **Revised duct inlet canopy**, with a

boundary-layer separator plate added.

The experimental Tiger first flew Jan. 30, 1954, before its fourth flight, which was a public demonstration (NAVY Aug. 16, 1954, p. 581) the plane had been supersonic under the control of project pilot C. H. (Corky) Meyer.

Bank of the Tiger resulted during an engineering gull testing run in January 1953. By April the Navy had written Grossman a letter of intent, and Douglas was underway. Assembly was well advanced by March 1954, and the first complete Tiger went from the experimental shop to the yards of the

larger firm on July 19. Production has continued since.



LATEST TIGER, redesignated F11F1, shows larger nose, second canopy and duct inlet and new vertical tail.

F11F Tiger



SPOILER UP, the Tiger rolls away (at right). Control system was developed for the Tiger FW-5 later adapted for the Tiger.

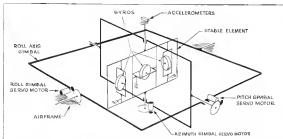
FENCELESS Tiger (below) was one stage of development of the fighter.

FENCED Tiger (bottom right) is now standard configuration, with fences extending only over part of the wing chord.



VENTED WING of rolling Tiger is created by mixed spools on upper surface of wing and restricted gaps on lower surface.





GYRO STABILIZED PLATFORM and linear accelerometers, shown schematically above, form heart of all known inertial guidance systems.

Inertial Navigation: Out of the Laboratory

By Philip Klase

Inertial guidance, the automatic navigation technique which will guide the unmanned vehicle safely to its target, is moving rapidly from the laboratory stage into operating hardware.

Although originally developed for use in long range missiles, such as the North American Navaho, and piloted aircraft, an adaptation of the inertial guidance technique may find future use in ocean surface and shore-based navigators.

Inertial type of guidance and navigation systems offers several significant advantages:

- Reliability to ensure precision.
- No ground facilities required.
- No satellite needed, preventing enemy detection.

With the present state of the art, an inertial system can weigh anywhere between 75 and 2,000 lb., depending upon its intended use, required accuracy and duration of the mission.

Although many of the basic inertial system components such as gyros, acceleration, compass and time systems, have long been used and should succeed, three major factors are a far cry from those required for inertial navigation systems.

The ultra-precision accuracy needed to give inertial guidance systems even tolerably acceptable performance is equivalent to the extreme precision accu-

racy found only in research laboratory standards. However, an inherent inertial guidance system must operate in the presence of temperature variations and shock encountered in high-speed aircraft and missiles.

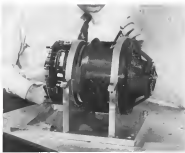
There are but little doubt that the art of inertial guidance instrumentation

has made significant strides in the past several years although details on specific devices are classified by military security. With an increasing amount of the nation's top scientific effort going into inertial development, even more progress can be expected in the next several years.

One evidence that inertial navigation is fast coming of age is the growing number of companies and organizations now active in a field which had not even seen over 10 years ago. Today, the list includes AC Spark Plug, Autonics, Bosch Avion Corp., Bell Aircraft Corp., Calsonic Institute of Technology's Jet Propulsion Laboratory, DeSoto Corp., Federal Techno-Scientific Laboratories, General Electric, General Precision Laboratories, Kollsman, Litton Industries, Massachusetts Institute of Technology, North American Aviation and Sperry Rand.

North America's Avionics Division and MIT's Instrumentation Division and MIT's Instrumentation Division De C S Dwyer, are two of the oldest groups in the inertial field. MIT generally is considered a fountainhead of know-how in the field, and several firms like AC Spark Plug and Massachusetts. However, work closely with Dwyer, conceiving MIT ideas into production hardware.

In addition, there are many other companies which are developing and/or producing inertial guidance sub-systems and devices, such as stabilized platforms,



STABILIZED PLATFORM, made by Edgemo-Pfister. Smaller units have been developed.

into Missile Systems

gyroscopes and accelerometers. Lee and Edgemo-Pfister are in this category.

An inertial navigation system is essentially a form of dead reckoning device. This means that the geographic position (latitude and longitude) or equivalent at both the starting point and destination must be known and must be set into the equipment.

It also means that inertial guidance cannot be employed against a moving target unless another type of target system is used in the final homing phase of the attack.

Over the information on starting point and destination has been inserted, the inertial system has been placed in operation, it is capable of determining and predicting the following information:

- Geographic position of the vehicle.
- Current velocity and track.
- Distance traveled and distance to destination.
- Direction to destination.
- Altitude of the aircraft or vehicle.

No Simple Dead Reckoner

With such information, it is relatively easy to design the system to guide inertial targets which can be tied to the aircraft or missile's automatic pilot to automatically fly to its destination.

Inertial guidance should not be confused with automatic dead reckoning (DR) computer, which have been in

operational use for some years. These devices provide a crude indication, but none of the information being merely different inputs. This type of DR computer is replaced by the AN/ASN-5 made by Rand Instrument Co. and another device made by Lee and Edgemo-Pfister Corp., Edgemo-Pfister and others.

The ASN-5 type of DR computer works from essentially the same information which a pilot or navigator would use if he were working out the problem manually. Airplane speed is obtained from a conventional speeded indicator, modified by external wind velocity which is set in manually as automatically by means of a Doppler or other type radar.

Automatic DR computers which employ Doppler/TM radar to measure ground speed are being produced by General Precision Laboratories (AN/

NAVIGATION, ACCELERATION AND INTEGRATION



ACCELEROMETER output is being integrated to determine distance vehicle travels.

ASN-5) and Ray Annetts (AN/ASN-67). The latter measures ground speed as interpreted in a function of time to provide aircraft displacement from the starting point, and this in turn is resolved into its north-south and east-west components.

Without such a table (ground speed) input, a DR computer is dependent upon an accurate source of velocity data (pressure or magnetic suction measurement). This is difficult to obtain on high-speed aircraft and missiles. Furthermore, accurate and continuous information on the velocity and direction of wind shift is practically impossible to obtain without external reference. An accurate drift or ground speed facility with the consequent design of pressure and detection.

The inertial navigators, on the other hand, substitute an entirely different technique of measuring aircraft drift or error, which eliminates the need for measuring indicated airspeed and wind velocity in drift.

Inertial Foundations

An inertial system is able to determine the displacement of the carrying vehicle from its starting point by measuring the accelerations of the vehicle relative to the earth. It is possible to measure these accelerations in terms of the forces that result in the vehicle and its inertial surroundings.

Velocity is the rate of change of distance with time, and acceleration is the rate of change of velocity with time. By a reverse process, if a vehicle's acceleration relative to the earth can be measured, a double integration of this acceleration over a time interval will give the total distance which the vehicle has traveled during that interval. Mathematically, this can be expressed:

$$\text{Dist} = \int \int a \, dt \, dt$$

Suppose for example, that an automobile starts from rest, accelerates and travels at 100 miles per hour for 10 minutes at a constant velocity. At the end of five seconds, the car will have covered a total of 125 feet and this will be resulting at a constant speed of 100 mph.

If the car were equipped with a device capable of measuring acceleration in the direction of the car's motion, consisting of a spring-retained mass in a glass tube, connected to its rear end pivoted by that "sensitivity" would be proportional to the car's acceleration.

If this signal were fed to an "integrator," a device which effectively multiplies the acceleration by its time of duration, the output signal from this integrator would be proportional to the car's velocity, or 50 ft/sec in the example cited. If this signal should be supplied to a voltmeter, suitably calibrated in feet, it would indicate the



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Basic Inertial Components

Accelerometers and integrators, sometimes combined into a single device, usually form the basic sensing elements of all inertial navigation systems. A vehicle dependent for use as an aircraft or missile, which has more than one degree of freedom of movement and which may fly at a variety of attitudes, involves much more complex and sophisticated than the simplified auto-rotate accelerometer and integrator shown. For example, an airborne inertial system usually requires:

- Two or three accelerometers to sense acceleration in the north-south direction, east-west direction and, in some applications, in the vertical direction. These accelerometers must be extremely accurate over a wide range of acceleration values, covering a range of "normal" landing thrusts to one, according to one report in the field.
- Associated with, and maintain a part of the accelerometer, is an integrator for summing up accelerations to a function of time.
- A pre-stabilized platform upon which the accelerometers are mounted, to maintain them in a horizontal position despite changes in vehicle position and attitude. In this respect, the stabilized platform can be viewed as a "passive gyroscope" whose attitude and heading is stabilized by a conventional sub-maneuver pilot.

Because an accelerometer cannot distinguish between vehicle accelerations (which it should measure) and g-force accelerations (which it should ignore), even a slight out-of-horizontal position of the accelerometer platform can introduce serious errors. The platform must also keep the accelerometers precisely oriented in azimuth relative to their

respective north-south and east-west axes or other coordinate system employed. Extreme high-performance systems require an accuracy in orientation of the platform stabilized to the required accuracy.

- A spherical-gyroscopic compass, capable of converting distance traveled into corresponding changes in latitude and longitude for the particular latitude of the accelerometer. For example, a distance of 50 nautical miles at the equator corresponds to one degree of longitude, but is equivalent to 1.1 degrees at a latitude of 45 degrees. The computer also functions to compute the distance and required heading which must be flown to the destination.
- Special corrections must be introduced to compensate for the fact that the earth isn't a perfect sphere, being somewhat flattened at its poles. The fact that a correction must be introduced for this difference in earth radius, which is only 13 miles at nearly 4,000, is one indication of the accuracy required in inertial guidance. Another correction is required for what is called "Casella effect" because the vehicle is rotating about a rotating earth.
- Other corrections may be required to maintain the gyroscopes for the vehicle's vertical velocity and height above the earth.

These then are the basic heading blocks of inertial navigation. They can be combined in a variety of ways, several of which will be considered in more detail in subsequent articles in this series.

Typical Airborne System

These heading blocks might be assembled into an airborne inertial system for long range aircraft which would operate in the following manner:

First, wind and static earth accelerations would be measured on a plat-

form which has three degrees of freedom (see sketch, p. 31). It can rotate about its gyroscope about the pitch axis, and about its yaw (azimuth) axis.

Two or three gyros, depending upon the type used, would be mounted on the platform to detect rotations of the platform about the roll, pitch and yaw axes.

When the plane's attitude or heading changes the gyro signals will sense zero across sensors to sense the appropriate platform pushed to its present displacement of the accelerometer from a horizontal plane.

To start up the system, the stabilized platform is oriented to the horizontal axis in azimuth so that the sensitive axis of the NS accelerometer is pointed northwards, which also orients the EW accelerometer. The latitude and longitude of the starting point and destination is first set into the system, and the integration is transmitted to zero.

As soon as the compass begins to take off, the accelerometers sense the resulting accelerations. Earth acceleration is integrated twice to provide distance covered in the NS and EW directions. These distances then are converted into corresponding changes in latitude and longitude and added to the starting point coordinates to show the plane's new position.

Spurious Gravity Acceleration

If the earth were flat, the system described would operate satisfactorily, as would be the case with the gyro reference. However, the fact that the earth is spherical, and that gyro attempt to maintain their position (and that of the platform) tilted in space causes a problem here to keep the platform always in a horizontal position as the plane or missile moves around the earth so that the accelerometers will not measure the spurious gravity acceleration.

A similar problem is encountered in a conventional inertial horizon gyro, which must maintain its axis vertical despite movement of the plane and the tumbling of the gyro to drift.

One problem is solved by using a small platform (or similar device) which senses its acceleration to determine the direction of vertical and to keep the gyro axis aligned in that direction.

However, as every pilot knows, when the airplane goes through turning maneuvers for an extended period of time, the horizon gyro displays a false reading indication. The reason is that during the turn, the gyro platform is subjected to centrifugal accelerations which cause it to "lean" that gravity (the vertical) has been displaced. The

gyro then aligns its axis now to the spurious vertical position of the platform, despite constant guidance can be used in short-legged aircraft, it is not satisfactory for long-legged planes and missiles.

Error Build-Up

Another serious problem which exists in the simplified inertial system described above is its build-up of error with time. Because computed distance is proportional to the squared power of time, a small error (about 100 ft) in the acceleration, even and in operation with sufficient accuracy to prevent an appreciable build-up of error during a long mission.

To solve this critical problem and the associated problems of maintaining the platform horizontal at all times, in critical systems designers have turned to an idea conceived more than 50 years ago by a German professor of applied mathematics, Dr. Mordechai Schuler. Schuler first described his idea in print in Physik, Zeitschrift in 1923, in an article entitled "Aberration of Pendulum and Gyroscope Instruments Due to Acceleration of the Transporting Craft". This principle now known as the "Schuler pendulum," "60 minute pendulum," or "earth's pendulum," forms the heart of all known inertial systems under development today.

The theory of the Schuler pendulum and its application to inertial systems, as well as the operation of a typical system employing that principle, will be described in the next article of this Airborne Warfare series.

FILTER CENTER

• Electronic Medusons - Ferrero agrees that the electronics business will mushroom in the next decade. But why?

• Double in five years, says General Electric's president, Ralph J. Centner. \$15 billion by 1968 and \$20 billion by 1975 is the prediction of Don G. Mitchell, chairman and president of Bell Telephone Equipment, Inc.

• New 500-ton Transairer Science-Tech Instruments test air "atmosphere" which senses air in motion to determine the direction of vertical and to keep the gyro axis aligned in that direction.

• Bigg's Memoranda - A high speed magnetic tape storage, which increases the capacity of the IBM 704 computer by 700% (32,768 words) has been announced by International Business Machines Corp. Research areas have to

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any size of the 36 lb wire requires only 12 microseconds with the new Type 718 magnetic core storage. All units in the storage unit are readily addressable from the 784, 1014 and 7802. Another similar type of magnetic core storage can be used to add 33,000 seven-bit words storage to the 15M 705.

► **Crystal Ball Polished Ore-Ds**, W. R. C. Baker, vice president in charge of General Electric Laboratories Division has come up with the following preview of things to come during 1956:

► **Microtrans** will become a more vital bit in the overall weapons concept of defense with greater emphasis on a



Dr. W. R. C. Baker

reliable control, guidance and interception of airborne weapons.

► **Translators** will make it more fault-tolerant in design for military, civilian and industrial use. New higher frequency and higher power translators can be expected.

► **Regulated resistor-capacitor assemblies** will be used more widely for constant resistable units and capacitors.

► **New subminiature assembly tube types** (AW Oct 17, 1955, p. 71) will be introduced.

► **No Government Inspection**—National Industrial Products Division of Holliston Electronics Corp., maker of tube sockets and terminal strips, has been granted permission to ship its components with out governmental inspection. The action followed a survey which showed that the outgoing quality level was "well within" the AQC standards established by the Chicago Air Procurement District.

► **New Noise Measurement Techniques**—The National Bureau of Standards reports it has developed a new method to obtain an objective, reproducible "noise index" for structural analysis subjected to shock, vibration or transient pressure. The new technique is being developed by the bureau's T. A. Pels and Air Force Department and Atlantic Flight Operations sponsoring.

NEW AVIONIC PRODUCTS

Components & Devices

► **Microtrans precision oscillator**, wire-wound, Type P, are specifically designed for use with printed wiring boards. Both leads project from one end and each require no support other than that obtained from leads. Resistor can



be operated at rated load at temperatures up to 125C, and each is capable of 0.1 to 1 watt, as available with resistances of 0.4 to 5 megohms, with tolerance of 0.05 to 1%. Resistor is printed on MIL-R-91A, Attachment No. 3. Resistor Products Co., 914 S. 14th St., Harrisburg, Pa.

► **Cable harness clamp**, Type NC1, consists of ring of braided wire, permits quick changes in wire routing. The Unifit clamp and keeper is



made of metal stem and can be operated at 250C. Clamp comes in range of sizes from 1/4 to 3/4 in. wide. Dakota Plastics Co., 260 N. Main Ave., Conger, Pa., Calif.

► **Battery airlock** with adjustable rate spring is now available in two basic models, the B21 and B25. Return spring torque can be set to very close limits. New bulletin gives mounting dimensions and standard spring rates. G. H. Leonard, Inc., 123 Webster St., Doris, Ohio.

► **Microtrans precision power resistor**, Type S, wire-wound, is suitable for slip-ring mounting which permits



up to 50% variation in power capability. Manufacture uses silicon coating to get allowed by high strengths of Bar metal coating. New Silicon coating are available with ratings of 2 to 10 watts in dimensions ranging from 0.075 to 0.75. Sign Electronics Corp., Rochester, N. Y.

► **High temperature resistor servomotor**, Type T1, is designed for maximum life of 1,000 hours of continuous operation at temperature of 150C, with average life of 150C. Motor consists of 1 1/2 in. diameter and weighs 3 1/2 oz. Input voltage is 315 v., stall torque is 0.4 oz-in., no load speed is 7,200 rpm., and rotor inertia is 1.7 gm-in.². Sign Electronics, Inc., Campbells Dr., 625 Main St., Waltham, N. Y.

Production Line Testers

► **Transistor tester**, Model T-82, checks the characteristics of point contact or junction-type transistors, low or high power (up to 50 watts), in either common-emitter or common-base connection. Tester provides direct readings of small signal parameters as well as static characteristics. All power supplies are included. Scientific Apparatus Corp., 5701 S. Union Street, Austin, TX, Mex.

► **Diode tester**, Model DT-300A, measures static characteristics of germanium and silicon diodes. Unit provides fan wind voltage of 0.05 to 2.0 v. d.c., up to 500 ma., and average voltage of 0 to 150 v. d.c., up to 3 ma. Reverse current has three meter positions for rapid checks at 10, 50, and 100 volts. An accuracy even permits double setting of constant temperature. Teletronics Laboratory, Inc., 54 Knicker St., Weehaven, L. I. N. Y.

Instrumentation

► **Pressure transducer**, Model P-1, magnetic resistance type, is available in ballistic ranges of 0.25 to 500 gpus, as in differential. Accuracy is quoted at 1% full scale. Unit is constructed with stainless steel housing, weighs 13 oz., measures 1 1/2 in. dia. x 2 1/4 in. long. Pace Engineering Co., 6914 Rock Ave., North Hollywood, Calif.

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0-1,000 psi, with a natural frequency of 5,000 cps. Minimum electrostatic sensitivity capability is 1% per 100G; in the 0-15 psi range. Wet sensors 1/4 in. dia. x 1/2 in. long, weigh 1/4 oz. Transducer output bridge circuit is built directly on an oval flat scale at 20 mils resolution, 5,000 cps carrier. Excitation frequency up to 20 Kc can be used. Galvanometer change rate impedance is 1% full scale/100K in Ω . Temperature range is -65F to 300F. Overall accuracy is better than 1% full scale. North American Instruments, Inc., 2420 North Lida Ave., Alhambra, Calif.

• Special-purpose receiver, Type 1670, available in two models covering the 57-260 mc. and 57-250 mc. bands, for use in telemonitoring, radioastronomy, space

and radio monitoring. Receiver has a 100-ke. bandwidth, discriminator bandwidth of 150 kc, maximum and a video frequency response of 20 cps to 100 Kc. All models have choice of fast or slow intermediate frequency control 75 or 10 Mc/sec. North American Instruments, Inc., 933 Jompson Drive, Skokie, Ill., 60076.

• Pulse-width laser analyzer, Model 126-A, is for pulse-width multi-coloring and direct recording of data onto a single track on an Ampex tape recorder. Model 126-A includes standard 600 sample/second pulse-width timer and a second analyzer on a single unit which is either compatible with the 732 or PWM/AM record strips in the new Ampex 600



Sioux airbone type recorder Unit weighs 11 lb. Applied Science Corp. of Princeton, P.O. Box 44, Princeton, N. J.

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• Analog-to-digital converter, available in 15 or 16-digit models, employs an epitaxially fabricated low-noise bipolar-ended glass diode. Shift angle is unipolar at rates up to 80,000/sec, and output signal is 5.7 volts. The 16-digit



encoder weighs 9 lb., measures 9 in. dia. x 5 1/2 in. high, requires special shaft, transmission, driving bars and paraphernalia. Application data is available from Baldwin Press Co., Encoder Div., 1589 Gilbert Ave., Cincinnati 2, Ohio.

PRODUCTION

New Materials for Metal Cutting Hold Promise of Production Gains

Two recent developments in cutting tools may bring production benefits to the system designer. Tools that combine hard, brittle, glass of alumina-type materials can be used at higher machining speeds than conventional tools, the ceramic have cut cost over containing sand holes without chattering or breaking down.

• Titanium diboride gives industries of being very successful for high-speed machining of hard high-temperature metals.

• Ceramic tools

The ceramic tooling development has been covered on the Norton Co., Worcester, Mass., in cooperation with the National Laboratories of Waterhouse, Amherst, Mass.

To date, Waterhouse's Alloy's tools have been used on SAE 4130 and 1020 steels, tool steels and brass. Pratt & Whitney Aircraft is investigating ceramic tools for tough, high-temperature metals that have proved difficult to machine with steel.

The Norton development based on either carbide or Garnet and Giant Britain, takes advantage of their properties when steels are exposed to conventional methods, while maintaining these when it is in service.

Some ceramic materials (such as aluminum oxide) are harder, have greater compressive strength and wear resistance than synthetic tool compositions. They provide strength, wear resistance, ductility and shock, and surface are generally superior to those of carbide and high-speed steels.

In tests at the Waterhouse Amherst, ceramic tools made of Norton's Alundum (aluminum oxide) have been used at speeds twice as high as those tolerated for conventional carbide, grade C6 and C7. A cutting speed in high to 1,600 feet per minute showed no detrimental effect on the ceramic tool, Norton says, and even higher speeds are anticipated where machine tools with greater power become available. Because of the high rate of stock removal, high powered equipment is needed—on one test run, a single-point ceramic tool drew its work at 93 ft/min.

use achieved by metallic tools under conventional conditions, Norton says. At maximum speeds and feeds, metal chips have been produced without chattering, the machine tool. Furthermore, the ceramic have cut cost over containing sand holes without chattering or breaking down.

Aluminum oxide occurs naturally in a very hard and dense form as corundum, sapphire and ruby. Mechanical properties approaching those of the actual forms can be obtained from material produced from a sapphire source.

Properties of the actual forms include compressive strength of 490,000-590,000 psi, tensile strength of 44,000-61,000 psi, tensile strength of 39,000-50,000 psi and cutting speed at 1,730 and 1,020 ft/min.

Methods of processing the aluminum oxide powder include hot pressing or pressing (either with no further operation or sintering) and cold pressing, slip casting, and sintering followed by sintering or firing.

For most efficient use of the ceramic tool, special attention has to be paid to toolholder design and tool design. The tool holder must give maximum support of the cutting tool, with clean-up areas evenly distributed to give near zero concentration. An optimal axial mechanical chip breaker must be used to protect against fluting.

Over-long work is recommended to reduce deflection and bending stress. Because of the compressed heat resistance of the aluminum ceramic tool, chips may be cleaned by grinding the work diameter to limit, Norton says. Reshaping may be done on the same toolholder design run which employed for sharpening ceramic tools.

The aluminum ceramic tool has an 80% life span.



SPECIAL HOLDER minimizes stress concentrations on ceramic cutting tool.

still in the development stage, Norton continues. Limited quantities are available to industry for research and machining studies.

Titanium Diboride

Read Development Corp., Cleveland, has taken a major step in the problem of applying titanium to cutting tools. The advantages of the synthetic material include very low heat; when used to machine high hardness steels, Read says. Tool life is up to 10 times that of tooling like carbide. Also, because titanium is harder than carbide, it is more resistant to erosion than tungsten carbide, Read believes it would be suitable for wear-resistant dies.

However, it appears that the material cannot be used for machining titanium. Because the metal reacts with the tool. Also, because of brittleness it is not suggested for machining rough castings or other rough-turned materials.

Thompson Products Co., Warren, Ohio, and Co. and General Milling Machine Co. have been testing the new tool.

Read made its first experimental metal-cutting tests corresponding to titanium in 1953, when the company was working on high-temperature turbine jet engine and jet airplane. Early tests showed extremely good cutting and wear properties, but the 1/2-in. diameter, wouldn't be an improvement to a steel blank and it was very brittle.

In 1955, Read has been trying to improve the material's ductility and reduce its brittleness, which is still something of a problem.

The basic raw material for the new tool—titanium diboride, boron and zinc—allowing elements are somewhat less expensive than tungsten and carbon, Read says. Furthermore, since the titanium diboride has about half the weight of carbide, tool life should be about half that of a comparable carbide cutting edge. Handling and finishing will not be more expensive than for carbide tools, except for reactions shown by the programs are needed.

To make the tools, titanium diboride and boron is powder form are blended in a tank and heated to 1,100C, at which point a self-propagating reaction sets in. During this reaction, titanium diboride is formed. This material is cooled, crushed to liberate particles size and mixed with the slivering materials, also in powder form.

The resulting mixture is specially blended and treated to make a cutting edge and is then made by pressing. The tool can be either sintered or hot pressed in a die.

Read has experienced good success in hot pressing the tool in carbon

molts. The mold, filled with titanium dioxide, is reduction heated to the required temperature, then pressed to shape, then broken away. Because of the high heat used (the mold has a melting point of 3,500C), the torch comes out with a "fingert" quality, needing no further work.

New Applications Seen For Ceramic Coatings

Our way of dealing with the problem of heat generated by supersonic aircraft and missiles is to make wider use of ceramics. W. M. Stern, lead engineer of the ceramic group of Boeing Seattle, proposes:

Among the applications Stern suggests:

- Heat protection of large skin areas and retrim internal structural members adjacent to powerplants
- Common protection for sections now protected by organic or chlorinated coatings which cannot withstand high temperatures
- Making areas where heat would destroy painted fittings

• Adhesive applications, because ceramic bonding will resist very high temperatures

Currently used organic coatings deteriorate at about 450F, the steel substrates, used in most jet engines, become inadequate at about 700-800F. Stern says, and also can treat hydrogen carbonaceous to many high strength steels. An opening into a plasma's skin at Mach 3 reaches a temperature of 600F. At Mach 4 the temperature is 1,500F.

Ceramic coatings will probably be some of the low-temperature-firing materials developed recently for aircraft. Stern says. He suggests further investigation of unconventional aircraft ceramic techniques, such as filamentary.

Some products that ceramics will see extensive use for protection against gaseous corrosion near powerplants, where temperatures reach the 1,300-1,500F range.

The ceramics are also expected to prove useful in protecting titanium alloys above 1,000F, where the metal becomes susceptible to embrittlement through reactions with nitrogen and

oxygen. Initial work shows that ceramic coatings will considerably prolong the life of a titanium component beyond that of a similar non-coated one.

Although development of coatings for titanium alloys will require considerable work, in Stern's opinion, the task will be eased because the typical aircraft metal elements can easily be applied to titanium alloys.

Coatings for aluminum and magnesium are expected to be good up to about 900F. Above that, the strength-weight ratios of the metals become too low.

Today's aircraft are hundreds of painted and stained surfaces. Pascolec coatings in various colors can take over where high temperatures make other methods impractical. Stern says ceramic paint might be applied to parts and then fixed with a lacquer, or flash-spensing through stencils may prove satisfactory.

In adhesive applications, ceramics comprise fibrous or flat bearing fire-light-temperature seal. They require little painting, impregnation and less external strength, compared with other materials available for similar jobs. Stern says these details may outweigh the disadvantage of a brittle ceramic bond.

Walkout Won't Affect Lockheed Expansion

Los Angeles-Lockheed Aircraft Corp. will continue to expand its research activities at its grandly vaulted division despite the walkout of 15 of its top members earlier this month over issue differences as opposed to how the company should run the division (ENR ENR 1/9 p. 34).

Regarding the division's plans, Dr. Louis N. Ridenour, new director of the Research Laboratories Group, had this to say:

"The division's company-sponsored program of basic research, exclusive of contract research, will be stepped up very appreciably. I might say, it is several hundred percent above 1953 program."

Ridenour contemplates no changes in the company's policy of administering its paid-for research program.

Inauguration earlier this month redefined the company's position on management of its research efforts. It placed them in a secondary role.

A majority of the departed scientists, according to Dr. Montgomery H. Johnson, have taken initial steps toward setting up their own research center in the Los Angeles area. He said the group approaches what a spectrum of talent which could be used in other than routine work. "It encompasses both the electronic and computers fields."



Supercharged B.E.25 Turbo-prop on Test Stand

Complete B.E.25 supercharged turbo-prop, designed to sustain a constant 4,000 hp from sea level to 20,000 ft, is now running on Avco Lycoming Co. test stand Six. B.E.25 has been ordered by British Overseas Airways Corp. for the 104th advanced version of the Britannia jetliner. Design studies for the engine began in 1952.



Exterior, Jettisonable Gas Load for C-123

Four exterior, jettisonable fuel tanks carry the engine gas load on Fairchild C-123B assault transports now being delivered to the Eight Air Force. In an aerial landing or other emergency, the pilot can release the fuel cells to reduce the danger of fire. Two tanks have been hit in the engine fuselage (over 700 psi), each gallon released causes tanks behind to flow and hang 450 psi gauge. In picture above, pilot has just released one of the tanks during a test run.

Navy Contracts

Following is a list of unclassified contracts for \$25,000 and over, as released by Navy Contracting Office:

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CAR Report on Pan American B377 Crash of Oregon

Stratocruiser Ditched When Engine Fails

A Pan American World Airways Boeing B377, N 9912V, was ditched in the Pacific Ocean, approximately 25 miles off the Oregon Coast, at 11:12; March 25, 1955, after No. 5 engine and propeller were lost and full fuel tank exhausted, 1,000 lbs. of fuel. All 35 occupants were evacuated but two fatalities and one serious injury occurred. The aircraft stall, after an estimated 10 minutes, as after the accident. The aircraft was ditched in the ocean.

Try 547276, a scheduled flight from Seattle, Tacoma, Washington, to San Francisco, California, Seattle Tacoma Airport at 10:57. The Portland, Oregon, and San Francisco stop. There were 13 passengers and a crew of 6 consisting of Captain H. S. Borden, First Officer A. C. Kinnison, Navigator M. J. Kerwick, Flight Engineer D. T. Fowler, Assistant Flight Engineer S. B. Clark, Purser Virginia R. Parks, Stewardess Elizabeth M. Thompson, and Steward J. D. Phipps.

The flight to Portland was normal in all respects with arrival at 9:16. Three days before, the aircraft had had a pre-flight inspection program. Only routine inspection and service was accomplished.

The flight left Portland at 10:30 and took off for Honolulu, Territory of Hawaii, at 10:51 on an IFR clearance. Weather was below that of minimum safe VFR. There were 13 passengers and the usual crew of 6. Takeoff gross weight was 139,494 pounds (maximum allowable was 147,036 pounds) and the center of gravity was located within limits.

The flight pilot was Col. Newberg and Navigator, Oregon, and thence to Honolulu to arrive at 10:58. Estimated flight time was 11 hours and 30 minutes. The flight completed over Newberg at 1:01 at 2,000 feet climbing, reaching 14,000 feet at approximately 1:10, and reported over Newport at 1:44 at 17,000 feet. The aircraft was then directed to make good the initial track of 225 degrees true.

Engine and Prop Troubles

Five-thirty minutes after takeoff, severe vibration occurred while cruising at 10,000 feet under VFR conditions. This lasted for two to eight seconds during which No. 5 engine and propeller were lost and 85 lbs from the aircraft.

The captain immediately disconnected the engine. Service control advised the crew to exit doors and the aircraft going to the right abut. At that point, the emergency "Manding" signal was sounded on both VHF and HF. Crew members in Portland was addressed by British Air Route Traffic Control.

The captain, on the left seat, tried to get the airplane under control. Attempt was made to climb 200 feet and going higher as the

"All times referred to herein are Pacific Standard and 45° based on the 24 hour day."

cloud the chances to keep the aircraft aloft. He still could not get the nose up, and the crew as though it developed was still an automatic spin. He tried the alternate trim tab and could not get her up, but the spin did not seem to stabilize. All 35 occupants were evacuated but two fatalities and one serious injury occurred. The aircraft stall, after an estimated 10 minutes, as after the accident.

Level altitude was regained by banking the left forward, and by use of the rudder and trim tabs. The aircraft was ditched at an angle of 15 degrees, butting downward approximately level, the aircraft ditched in the ocean.

Attempts to get rated engines were made and a message was broadcast that ditching was imminent. This message was sent at approximately 1:45.

Preparation for Ditching

The aircraft was ditched in the ocean at 1:44. The ditching was successful. Crew members following the emergency, survival kit, and emergency instructions for ditching. All passengers had been advised as the type deck of the aircraft was not fully inflated and life jackets donned.

The aircraft touched down under control in a shallow bay with water level and the aircraft ditched in the ocean. The aircraft was ditched in the ocean at 1:44.

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SAFETY

One other passenger later died in a self-burn response suit shock.

The purser, a woman, although wearing knee shock boots and leaved for only severely injured passengers to the second deck, where 750 feet altitude.

The cause of ditching was determined as 1:12 and the position at lat. 49° 57' N; long. 121° 17' W. W. approximately 15 miles off the Oregon Coast. The aircraft was ditched in the ocean at 1:44.

INVESTIGATION

Investigators declared that the ditching of the flight was an full conformity with all government and company requirements. Up to the point of an entire the aircraft was Portland, Oregon.

Water approximately one mile deep produced injuries of the wreckage. Small light injuries were not significant as 15000 feet was regained only water shock on the coast.

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Possible Causes

List of the engine and propeller could have been as follows:
(1) Failure of the engine mount
(2) Failure of the engine mount
(3) An unditching propeller caused by the failure of a blade.

Since the engine and propeller could not be recovered, there was no opportunity to examine them.

The first possibility—failure of the engine mount—was not only a temporary solution that when the engine fell it took its second with it leaving nothing forward of the aircraft except seat parts such as water and fuel.

Regarding the second possibility—that of engine mount collapse or removal—there were no other passengers of the crew. The aircraft was ditched in the ocean at 1:44. The ditching was successful. Crew members following the emergency, survival kit, and emergency instructions for ditching. All passengers had been advised as the type deck of the aircraft was not fully inflated and life jackets donned.

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pinion declared that there have been no known cases in which an engine has been fired from the cockpit as a result of sudden stoppage.

The third problem—a substantial propeller speed fluctuation in the case of the engine. Other blade failures of this type-progressive deterioration have produced the results.

There have been five previous instances of total propeller stoppage from the cockpit and use of partial apparatus. Of that five, two were definitely caused by propeller blade failure, and the remaining two failures engine were not necessarily unduly related from the same cause.

Proves Propeller Features

During the run-in of the propeller, the manufacturer developed modifications and more extensive inspection and maintenance procedures, all of which were aimed at improving the safety of the engine. The most recent modifications was to seal-plate the blade ends to minimize unsecured ends and gaps. Studies on the FAA flight of B-37 aircraft were completed and maintained in accordance with the manufacturer's best service instructions.

The investigation of the accident is related to a check of the results of inspection to determine if there was sufficient inspection to determine the extent of blade failure. These inspections included X-ray, magnetic and detailed visual examination of blades externally in areas not easily visible, i.e., outer rubber lagging. Heretofore this area had not been tested.

The comprehensive program included tests and gauges around the gauge around, a constructional inspection following flight. Furthermore, as the program evolved, a revised blade, not manufactured as demanded on the aircraft at that time. Work of this kind revealed that it resulted from fatigue and that it originated at a rounded area under the rubber lagging.

On March 25, 1970, following the accident no crack had been reported as being found on a metal plate blade. However, the inspection requirements that were applied to the model B-37 blades that were associated with corrosion and not blade failure from the same cause while undergoing fatigue testing at the factory. The X-ray program revealed one new blade of B-37 cracked beneath the rubber lag.

The crack had occurred during static, moisture tests and not during fatigue testing.

Corrosion which is known often to serve as the first bridge before was found on 137 of the B-37A/Pacific Model Two blades. The routine airframe service checks assess on the 144 at Fort Worth consisted of a visual inspection of propellers, landing gear which, two control surfaces, engine cowling etc. The inspection was made by two mechanics per examining the two bleed air ducts which are attached to the r/t. No imperfections were found.

Propeller Speed Control

The propeller speeds of the aircraft were externally controllable. Control could be externally as well as internally. The electrical system supplies fuel for

the low indicated engine speeds and was to control blades, both of which are mounted to all five engines. One master control handle in the instrument case is the master control and the second one is a circuit for manually selecting engine # 1.

In this instance, the having five of No. 1 engine obviously created a short in that portion of the system serving No. 1 engine. A subsequent check of the instrument case to increase r/p by 10 in use of all engines simultaneously (the actual propeller speed) and operating of the master control handle in this r/p in all cases of the remaining three engines could be changed.

Continuing indicates that the engine check the taxi location and upon it stopped successfully to get instantaneous action of a r/p in this time the aircraft was ready to fly. The captain stated that the r/p is now revised.

Unforeseen Contingency

Flight Engineer Freck attended his duties in 1970 on propeller control cases. It has been established, however, that the specific conditions that occurred in the accident was never taught in any of these areas nor had the company used any specific instruction in regard thereto. No could the entire type of engine be approximately present in the General Flight engine available course.

Flight Engineer Freck had been retrained by FAAA from December 1970. After serving in various capacities he was placed in a flight engineer in April 1971. Mr. Freck had served on flight engineer as B-37 aircraft used 450 hours.

Assistant Flight Engineer Freck, who was accepting the jump seat at the time of the emergency, had a visual check before the engine and observed that the anomaly below was visible to that previously described. It was approximately 100 feet and delivery of the engine and engine in controlling the aircraft. He recalled that the three engines were running smoothly.

In the post flight check reported to his power factor that the pilot could have had difficulty in raising the right wing if he would get these three propeller flow. However, the flight engineer had no trouble to get one r/p in engine. Backlash then occurred over and advanced No. 4 electric control system had been observed the airplane heading to be 600 feet.

When r/t had been set, a gear check was to provide extra horsepower, speed, and to prevent accidents from taking on the water. The origin changes had not been set, and the propeller rate and control location. Backlash then set a hand pump engaged in the r/t.

Three Engine Performance

The manufacturer of the aircraft had provided performance charts for three engine flights of the Boeing 77. These particular data curves were based upon actual flight test and wind tunnel data of all the aircraft to verify flight conditions that existed below the loss of No. 4 engine flow a water r/t.

The chart depicts the capability of the

would aircraft with No. 3 trim fuel, as happened in the instant case. They show, varying loading per sq. ft. per centimeter of diameter, in the instrument case is the required maximum power requirements that the aircraft require 1400 gross weight to fly at a stall speed, 1700 p.p.h. (111,000 pswh) it is, required gross weight also stipulated at the No. 1 power package.

However, according to the manufacturer's present data there being an actual deviation of the result to be an actual gross weight of 1400 p.p.h. (111,000 pswh) it is, required gross weight also stipulated at the No. 1 power package.

Captain Freck testified that while in 4000 feet he was not sufficient time to consider changing fuel.

Life-Raft Launching

The three 20-man life rafts were loaded with 15, 1 and 2 occupants. One crew member and one crew member in one raft. In the second were the second crew member and first purser, and the third raft had the captain, second officer, crew attendant, and purser to make a total load of 15. The aircraft r/t to the first purser which were not record was intended to be 100 feet or more. The heading was in all three rafts was 21 knot long.

Two of the life rafts were of one make, the other of another. The first two had defects and some limited instructions in structure but not than had the third. Crew members were not to be in the rafts. The rafts were not to be used in the water. The rafts were not to be used in the water. The rafts were not to be used in the water. The rafts were not to be used in the water.

The rafts had been set, a gear check was to provide extra horsepower, speed, and to prevent accidents from taking on the water. The origin changes had not been set, and the propeller rate and control location. Backlash then set a hand pump engaged in the r/t.

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The equipment being containing fuel, varying from 10 to 100 lbs. of fuel, and the r/t was to be used in the water. The rafts were not to be used in the water. The rafts were not to be used in the water.

Evacuation Was Overtly

Both programs and crew members noted that the evacuation was orderly and

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"Woody" Woodward can "see" storm centers 150 miles away... for his DC-8 is the first modern-equipped executive aircraft in the world! He flies safely when other planes are grounded. Woody says, "My kind of all-weather flying demands the best possible engine overhaul. Naturally, I use Airwork overhauled engines."

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This generator can be furnished with either square or round mounting flange and is suitable for use on Continental, RVA, P. A. W. 961, P. A. W. 1210, P. A. W. 1216 and Wright CR 122B engines.

For auxiliary applications, the generator conforms to AF Drawing 522 0598 and will deliver full output at 35,000 feet altitude and 75% rated power at 50,000 feet.

CHARACTERISTICS

Rated Max. Speed for Regulator	2000 rpm
Rated Max. Speed for Regulator	2000 rpm
Rated Power	300 watts
Rated Weight	31.4 lb.

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completed in an expenditure quarter, with the exception of the difficulties heretofore mentioned.

The American had an established Avrofit Emergency Equipment Training Center to train instructors. First to take the heavy guns, equipment training all right, planes and land lines completed a course of an eight month course, including a complete knowledge of the aircraft's design, most parts, etc. etc.

The flight was under instruction, of ground take. A plot of an aircraft you have outlined the crew's technique, so the aircraft's superior while depending. It shows, eventually, that student, from crossing altitude of 50,000 feet on a north westerly course, the aircraft made a full 180 degree turn to its right and then turned right, slightly about 150 degrees, also to its right, and was then left to the side as it went below 500 feet (just past to drizzle).

The direction of drizzle was about opposite that of the main crossing flight, about a full turn and a half to the right was made between the start of the turn and the drizzle; the wind stopped once was recorded in case records.

ANALYSIS

In the consideration of the accident and its results there are a number of factors, which are discussed as follows:

- (1) Initial failure
- (2) Control difficulty and its possible cause
- (3) Inability to recover; i. e. in other three stages
- (4) Non-landing of fuel
- (5) Drilling, removal and cause
- (6) Corrective action

Initial Failure

The elements which occurred immediately before No. 1 power package started are indicated in a timing picture of losses, propeller blade failure.

Despite the power package not being recoverable for three or four minutes to doubt that the trouble was due to blade failure. This belief is based on the known history and subsequent examination of model 2017 blades. The examination disclosed before mentioned common fault—metal failure during the drizzle of 15.15, at 43 of the first 513 blades. One of these 43 was cracked.

Examination of blades in other stages revealed one cracked on a delivery aircraft and one cracked on another in contact aircraft.

The same blade is characterized by its stress history; hence to steel of similar regularity. The fact denotes subsurface inspection during manufacture and while in service. Corrective action is now under way, as will be shown later in this report.

Control Difficulty

The flight is unable to determine the reason for the initial control difficulty.

If not, but here appeared by an undeterminable irregularity of the line, but the irregularity covered in the first, but plain, only of No. 1 engine.

It may also have been due to some de-



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bonance of the vehicle, particularly at the emergency, caused by some sudden wing aspect or dipole from the No. 3 power engine. This possibility is strongly ruled by the difficulty the pilot had in making the take-off accounted for calling off the first officer for assistance. However, the impact at the time of it, immediately after the failure was noted by my assistant.

Whatever the cause, shivers control was required after rapid loss of considerable altitude and the aircraft was ditched and crashed.

Inability to increase RPM

The flight engineer on duty at the time of the accident did not verify. Consequently, the nature of the difficulty that he experienced in attempting to increase engine speed can only be learned from other testimony. The testimony is largely that of its control flight engineer and, to a lesser extent, of the captain.

There was no evidence that there was mechanical or electrical impairment of the control system of Nos. 1, 2, and 4 propellers.

There had been no trouble of any sort prior to the emergency.

In view of the known characteristics of the propeller devices in the propeller control system, it can be concluded that the malfunction operation of the propeller control was the responsible factor in not getting increased r p m.

On June 21, at approximately April 20, 1951, the 14 prop magnetic control lever was replaced by wires using 3 amp thermal type circuit breakers in both main circuits and the 5 amp fuses in the individual circuits were replaced by 3 amp fuses.

This change altered the flow associated with the malfunctioning circuit to indicate that there the remaining circuit was affected. This modification as detailed in Hamilton Standard Service Bulletin No. 28-23 entitled "Block-assembly Type-3 Servo Circuit Protection," dated December 29, 1951.

Compliance with this bulletin was not mandatory by the CAA although the air partner of its test was effectively denied by the circumstances of this test event and it was made mandatory by the CAA, April 21, 1951.

It may be pointed out that this modification was also applicable to the engine's fuel oil "Dodge" DC fuel and had been made on them, it was the company's intent to make these modifications as its part of Boeing 375's as soon as practicable.

Non-Dumping of Fuel

As previously stated, emergency operation of the subject aircraft would have been feasible with No. 3 engine gear at 2,040 r p m had no weight loss related to gear at 1,718.000. The weight loss would have exceeded a weight reduction of approximately 11,000 pounds.

The maximum rate of fuel flow during dumping at 151 indicated RPM is approximately 1,163 pounds per minute. This is nearly 1,163 pounds more than the flow would be for 1,125 RPM, from an

and difficulty in detecting an approximately one minute.

It appears that if fuel dumping could have been started immediately after the failure the aircraft could have been kept level enough to land back water nearby on the three good engines. However, the captain's mind was occupied in attempting to control the aircraft and the problem of the inability to increase r p m of the engines, the captain stated that there was not sufficient time to even consider dumping fuel.

Ditching

The aircraft was ditched in daylight under sea level conditions. This was the first ditching of a civil B-377 aircraft, consequently, there was no direct knowledge of ditching characteristics. The aircraft sustained about for approximately 20 minutes.

Under these favorable circumstances and with comparatively few passengers (31) it might be anticipated that little difficulty would be experienced in getting everyone aboard his life. Such was not the case. The various means the life were cut short after being launched as explained. The uncontrolled seat occupied getting into the water and swimming by the life which were being dented by the light wind. Two of the crew members and one passenger were unable to reach them and perished.

The conduct of individual passengers and

individual crew members was high. Peter Nardis, Radio displayed courage and devotion in doing whatever was necessary. However, the fact that the life had to be used and unavailability made it extremely difficult to stretch all persons and was a contributing factor to the loss of life.

Search and Rescue facilities were put into effect promptly and efficiently.

Corrective Action

The occurrence and investigation of the accident resulted in a number of corrective measures being initiated, among which are:

- (1) Life rafts were stored more securely.
- (2) Additional equipment of the propeller blades were required and the proof between previously required inspections was at more intervals shortened.
- (3) The schedule of maintenance of propeller block indicator devices which had just been brought down in view the crew of an approaching blade failure was expedited and then crew made mandatory by the CAA on July 30, 1951.
- (4) The manufacturer revised development of a solid structure propeller blade for use on B-377 aircraft. The CAA recommended to the CAA that all Hamilton Standard B-377 hollow steel blades be removed from service on the B-377 aircraft at the earliest possible date consistent with the manufacturer's ability to supply such better blades.
- (5) The Administrator, by letter dated



X-Ray Propeller Inspection

Pan American World Airways uses a portable X-ray machine to inspect propellers (Boeing Stratoliner) and shows in photo, rollers, nozzle bearings, impact, landing gear and fuselage structures. Since PAA started its prop inspection program, more than 4,800 X-ray shots have been made. Gates area of each blade is examined every 1,600 to and the blades themselves every 2,500 to. Edith Anderson, a physicist in line crew of X-ray equipment (AIP Sept. 28, 1954, p. 71), is now employing the technique on DC-7 wing structure. Two men are needed to support the structure from wing station 411 to 411 inch. Edith estimates it takes 1,600 man hours a year in her area to open wing inspection shots. The Ames and systems of any 110 or in 1200 weight without special treatment. It is possible from Edger Anderson, Inc., San Francisco, Calif.

BALANCE—Forecast of Northrop achievements as an efficient organization will be shared between the poles of advanced research and economical production. Northrop researchers in physics, aerodynamics, metallurgy, electronics, aerodynamics, optics and many other fields are doing their work year in advance of the need. From this long-range thinking and planning within the Northrop complex have come the Radisson-Corsair's family of gliders done and made, first and foremost in their field; the Scorpion F-49 intercepter, present defenders of our heartland approaches; and Strik 52-62, deadly intercontinental A-bomb carrier. As streamlined as its products, the well-balanced Northrop organization is at work on even greater weapons to strengthen this nation's defense... and is more ready than ever to develop and produce them efficiently and on schedule.



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American World Airways and was recently considered by the Civil Aeronautics Administration. The aircraft was made ready May 18, 1949, and had accumulated approximately 11,657 hours. It was equipped with four Pratt & Whitney R-4600-16 engines and two Hamilton Standard model 2410 propellers with model E111723 HW blades.

The engine and fuel systems and propellers had been maintained in full compliance with prescribed methods and within all time limitations.

New Life Vest for Ditching Survivors

A life-line-equipped life jacket which can be thrown 50 ft. to rescuing survivors of a ditched aircraft has been developed and successfully tested by Pacific Northern Airline, Inc.

Tests with the Navy and Coast Guard showed that a neoprene standing in an inflatable rubber mat could throw the vest about 50 ft. It is best to throw the mat inflated in all but a very strong wind. Navy personnel said the mat is easy to grab hold of and get into while in the water.

The mat is made up of one of the two air chambers of a standard "Mat



West life preserver. Some 75 ft. of line which can be attached to the C-47 inflation cartridge (shown) which allows the vest to be inflated by giving the cord a sharp tug. Ring and cord can be folded into a package measuring 6 inches 3 by 10 in. dimensions and weighing 11 oz. Carriage webbing horns are provided—two for the lines and a third to hold the survival out of the water. Total cost is about \$2.65.

Noted as desirable, retrievable life vests was reported up by a recent Senate census dishing out Oregon (See p. 45) survivors as liferafts were unable to help others struggling in the water because they had nothing to throw to them and the sand was drifting off the raft's mat; later this an average person could swim.

PNA says that it will furnish free drawings and specifications of its three-chamber life vest to anyone who is interested. Address: Seattle-Tacoma Airport, Seattle 55, Washington. Also: Frank Hoover, chief pilot.

WHO'S WHERE

(Continued from page 15)

R. L. Collier, manager of military repair units department, Alcoa Division, General Motors Corp.

Beverly W. Duncan, in charge of research and development, Miles Pattern Casting Co., Wichita 3, Mo.

Richard E. Williamson, military service representative, Co. Inc. Chemical Co., Inc., Denver, Colo.; formerly Deputy Chief, Jackson & Manaster, Research Development Division, Minneapolis, Air Mutual Company.

Michael Goldstein, sales manager, Alloy Products Company, Cleveland 10, Ohio. He was formerly president of Control Car Co., N. Y. C.

John F. Griffin, manager, Electronic Material Development Dept., Science Products Department, General Electric Co., Westfield, N. Y.; **Edward A. DeMotte**, general sales representative, General Motors; **The Department Store**, N. Y. C.; **Drew Co. Inc.**, product planning and market research, section, General Department, Detroit, Mich.

Paul L. Hansen, director of sales and marketing, Tridair, Inc., North Wales, Pa.; **James A. Gilbert**, in charge of metal and sales research and **W. Aldert**, in charge of advertising and sales promotion.

Capt. J. E. Stone, British Overseas Airways Corp. sales manager for Northwest Air, replacing **L. F. Wood**, transferred to London.

Douglas Roberts, advertising manager, Co. Indian Pacific Airlines.

Frank B. Brady has taken a major part working as Mills, Peterson & Mills, Wash. region; **D. C.**, architectural and engineering firm. He was formerly project engineer, Air Transport Assn.

Thomas Harrell, product representation manager and **Donald J. Wink**, aviation chief engineer, Chas. E. Jones Division, Pratt & Whitney Co. Inc., W. Hartford, Conn.

Kenneth B. Luskis, administrative assistant to vice president, Industrial Manufacturing Co., W. Des Moines, Ia., supplier of all engine models to Pratt & Whitney Aircraft Division, United Aircraft Corp.

Norman S. Brundage, aviation sales mgr., University of Southern California, Los Angeles, where he will conduct an aircraft accident investigation and prevention course.

Allen O. Gerloff, project director, National Secretary of Scientific and Technical Personnel, National Science Foundation. He was formerly with National Aeronautics Committee for Aeronautics.

Robert H. King, assistant chief manager, Kaiser Manufacturing Co., Denver, Colo.

William H. DeBorja, plant operations manager, Ingersoll Rand, Thomas A. Edison, Inc., W. Orange, N. J.

Dr. Wilbur B. Storer, editor, Journal of the Aeronautical Sciences, Institute of the Aeronautical Sciences, N. Y.; formerly director of the graduate school of aeronautical engineering, Cornell University, Ithaca, N. Y.

Walter J. Hapgood, sales engineer, Fairchild Manufacturing Corp.'s Metals & Fabrication Division, Cleveland 10, Ohio.

William J. McGeehin, sales promotion manager, Libbey, Inc., Glendale, Calif.



IN THE AIR The plane has a top speed of about 63 mph, lands at 25 mph. Gross weight is approximately 400 lb. It takes about a half hour to convert the plane for road travel.

Home-Built Roadable Plane Costs \$250



WINGS FOLDED, lower wings, roadable rolls along the highway powered by a rubber-mounted 40 hp Continental. Builder Daveo Bryon plane requires no-pilot license for road use.



WINGS EXTENDED, Bryon roadable starts takeoff. Costing \$250, plane took 15 hours to build. It was easily salvaged from a plane; tail and engine are from a Cessna Ruby A-6.

Sheer Poetry

We write the "poets" of today.

Apparently they make first page news. They also command attention in industrial publications. Industry-wide they contribute the warm glow of experience. And industry, in bringing their heavenly ideas down to earth, stands aside there to share the spotlight.

Though these modern poets do not write in verse, they appeal to our imagination. To us, their communications are essential. For theirs is the sheer poetry of wordless days of accomplishment. They themselves are the engineers and scientists who are the uninspiring of our age. We read them, one and all, and urge them never to cease in their quest for awe-inspiring achievement.

At Sikorsky we have poets like these. We need more. Are you one? If so, please write to Mr. Richard Atzer, Personnel Department.

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Transpacific Travel Sets Record; Tourist Fares Behind Increase

By Brian Long

Los Angeles—Summer passenger traffic across the Pacific this year surpassed all expectations, with tourist-class flights registering the largest gains.

In exceeding the already optimistic predictions, Feb and August passenger travel registered an increase of more than one third over the peak months of 1954 and exceeded for the August month the three transpacific U.S. scheduled airlines have had to date.

Without exception, tourist class travel represented the major portion of the recent travel which included these items:

- Pan American World Airways recorded gains over last year of up to 35% during the summer months, 10% above the anticipated total increase.
- United Air Lines gained 25% over a year ago, 15% beyond predictions.
- Northwest Orient Airlines production was below last year's at 37% gain.

Proportion by the airlines for still greater increases was assured already by the summer months of 1954. Traffic and sales manager of Pan American's Pacific Division, W. E. Miller, "We are planning on the biggest tourist season in the history of transpacific air in the Pacific, not only in Hawaii but all over."

- The past season's high volume losses can be attributed generally to these factors:
 - Many conversions to tourist service.
 - Straggling production programs.
 - Fuel rate hikes.
 - Newer equipment.

All of these factors are expected to contribute greatly to 1955 increases.

Fee American

A total of 17,237 Pacific coast passengers were flown to Pan American Pacific ports during July, an increase of 73% over the 11,544 flown in the same month of 1954. In August, the total was 18,015, 22% more than the 15,301 for August, 1954.

Pan American's greatest increase was in status to and from the Orient, with increases in both Hawaii and Hawaii to and from Tokyo, Hong Kong and Manila. Transpacific traffic to the Orient, as compared with the same months of 1954, rose 141% in Feb and 275% in August.

Fares to the Orient to the U.S., it jumped 39% in July and 50% in August. South Pacific and Hawaii-to traffic also showed substantial gains.

During the summer, Pan American's route between Hawaii and the Pacific

also made of Super-G Constellation, N.W.A. now offers 10 separate services across the Pacific each way. An added local weekly round trip will begin on July 1. Another factor was the introduction of new type aircraft.

An added factor was the early season, Northwest and, was movement of several forces personnel and dependents by commercial airlines from Japan, because of the shortage of space in military transport ships.

Wants Promotion

Concerned with the introduction of new Super Constellation, Northwest stepped up the number of promotions of its crew. "We should be able to project such increases further into the new year."

Concerning the DC 6Bs and DC-7Cs, averaging to 342 airlines, are scheduled for delivery within the next two years. Donald W. Noyce, Northwest's president, recently announced that the airline will decide, probably within the next six months of 1955, to lease but two or three DC-7Cs, to be replaced by either Boeing or Douglas.

Although the DC-7Cs will not go into service until 1957, Northwest now it has effectively increased its seating capacity by upgrading Shasta aircraft at the top of the Alameda Island chain as a refueling base for flights to and from Japan. Thirty deliveries in 960 units shortly from the previous Gold Key, Alameda to Seattle, Alameda and Seattle an increase in payload of about 60% and an increase in number of seats of about 15%.

Replacing Shastas under a 10-year lease with the Civil Aeronautics Administration's new Constellation Northwest is the first American carrier ever to operate this new aircraft. Also, it is the first commercial purchaser of a Golden GCA jet in Seattle, Alameda and Honolulu.

Present schedules here, Northwest is flying one-to-two between Japan and the U.S. (via the Shikoku route) twice a week. The scheduled alternate flight will go via Anchorage, Alaska, and the route of the refueling base, due to the presence of the new Constellation.

Pacific Traffic Increases

	Tourist		First-Class		Class
	July-August	Gain	July-August	'55	
U. S. & Pacific	6,286	7,944	2,658	26,515	28.9
U. S. & Hawaii	893	914	275	969	10.6
U. S. Orient	5,393	4,956	2,383	1,546	56.6

This table includes U.S. carriers only. Japan includes any air peak season in August October, and reports a gain of 53%. During these three months of 1955, it carried 3,303 passengers between Tokyo and Honolulu and U.S., compared with 2,133 in 1954.

Inadequate Aids Bog Transoceanic Traffic

By R. C. Hanson*

New York—Inadequate air navigation and communication facilities are seriously hampering North Atlantic air traffic and the situation is steadily worsening.

North Atlantic enroute aids as high as 3,000 a month. There are at any one time during the day about 75 aircraft en route between Canada, Newfoundland and Bermuda, including Canadian and airline owned 210,000 passengers between Europe and the U.S. in 1948. This will carry 780,000 passengers in 1955. Practically all that by 1960 there will be 1 million passengers through the Atlantic.

There are three major factors causing congestion across the 2,000-mile north-south route:

- Inadequate navigation aids.
- Navigation difficulties.
- Aircraft performance.

Communications

There is a complete absence of direct communications. Communications of enroute are put on the local teletype circuit and routed to the various communication centers where they are accepted for transmission across the Atlantic by radio teletype.

Control Clogged

The communication's center is one of the best bottlenecks in the present. All enroute traffic is handled through a single station, but radio teletype facilities are not as well developed. These reports receive 18 elements of information, including a complete weather forecast in 10 or 15 seconds, position, altitude and height. In addition to these reports and traffic elements, various company messages are sent over the same wire which increase the message load.

Climate, message and position reports are only two of the several forms of information on each flight that also through the communication's center. There also is a departure message when the plane is airborne. It is the responsibility of the originating airport to monitor each flight and then pass it on a message to the enroute point when the aircraft arrives at its destination.

The result, says Hanson, is that a few minutes, but sometimes hours late. There are cases where the message has been received the day after the arrival of the aircraft.

It is not unusual for a routine clearance request to be late before it is received.

delivered. The time is estimated by the minutes of increasing the message to send from the origin, the communication center and the traffic controller.

Only the original route has the authority to cancel a message. In many cases a flight will reach its destination before receiving an answer to a flight report. Although it is obvious that the information is useless after the flight is enroute, the message still continues through the process.

Fuel Alerts

Even if the radio teletype circuit were to operate a full speed around the clock the problem would be to get clear. There are numerous breaks in transmission in the air north, due to radio propagation difficulties, some lasting as long as 10 hours. When the circuits are closed, the aircraft are forced to use the Morse land law which is a time-consuming method.

Propagation difficulties have been noted as impossible for an aircraft to position itself. If an aircraft is interrupted, the center must be alert for the "outgoing" aircraft. The Center center in 1951 issued more than 400 emergency alerts in 53% of the alerts, the center was forced to act in many of the aircraft to make contact.

Each alert consumes 30 minutes of the controller's time. Another message an further delay.

Each gain through the Atlantic is interrupted, the center must be alert for the "outgoing" aircraft. The Center center in 1951 issued more than 400 emergency alerts in 53% of the alerts, the center was forced to act in many of the aircraft to make contact.

Insured Transmission

Another complaint to the traffic control center is the device known as the North Atlantic oceanic. Military planes usually make direct contact with the headquarters through military frequencies. The North Atlantic oceanic planes are completely and must be returned back.

In some cases an aircraft is unable to contact a land station, but is able to contact a U.S. Navy or Coast Guard cutter in the North Atlantic. Messages to Washington and then be an untransmitted to the proper destination.

tion across the Atlantic. The work of test and aviation engineers.

A typical aircraft in the assignment of an aircraft operating its program to South America to an altitude of 17,000 ft.

Although the aircraft has usually received a green sector, the center can not use the same space as it have received a report from the aircraft. This causes large areas of recent airspace to be unused while the aircraft is in the process of transmission.

Center have worked out a plan to combat this situation by making local assignments for the assignment of certain blocks of altitude to each other. The altitude program Center now is 13-14-15,000 ft. While Shasta flies 16-17-18-19,000 ft. This permits more positive control and speech clear areas. More attention to this plan is that each center can not need exactly four levels.

Space Assignments

To cope with the uncertainty of position reports, the center should alert pilots it is necessary to assign each flight large blocks of airspace. Reports that is provided by altitudes 120 miles for level flying, a minimum of 1,000 ft vertical spacing and up to 40 statute miles.

Each center keeps track of individual flights by marking off every 5 deg. of longitude and using these lines for plotting aircraft tracks.

Another problem is the system is the fact that aircraft do not necessarily give their position or their estimated position in their 5 deg. box. An aircraft may estimate its next position at 32 miles from the assigned position, but then must estimate to post the 5x at 30 deg. This occurs on numerous flights and the center spends considerable time working flight plans and out-of-range positions.

Navigation Aids

There is no "oceanic return" long-range navigation aid used by the North Atlantic oceanic. Some use external navigation. Others use Loran or Goral and other external. There is no precise tracks or information can exist.

The navigation aid problem is two-fold:

- Long-range aid requirement.
 - Land-based aid requirement.
- A major requirement is a long-range aid that will provide an adequate number of precise over-ocean tracks accurate enough to be used by the aircraft in a lateral signal. This system also must provide for a sufficient number of ac-

crete lines along these tracks so that ground programs can be decentralized without substituting which reduces the required longitudinal spacing.

Even greater precision is required for the navigation aids at terminal areas as well as a greater number of aids. The cost terminals have good equipment, but the need is within the 150-mile arc. The lines are needed to serve as holdover points at "gateways" to outer areas, to clear obstructions and to facilitate access to various altitude levels.

At the present time there is no framework of defined tracks and fixes upon which a more accurate system could be built. Until this framework is established through a standard system of long range navigation, it is necessary to continue the practice of installing large blocks of equipment to each flight.

Aircraft Performance

Aircraft operating on the transatlantic routes have their own special requirements for long range flight. No one can cope with a standard, or preplanned, traffic pattern across the ocean. Each flight is a separate case and an aerial control must fit itself to the actual situation.

Fact is the principal problem. Each plane has an altitude at which it most operates in order to obtain the best performance.

There are three general classes of aircraft operating in transatlantic traffic:

- **Updated planes which operate at altitudes below 10,000 ft.**
- **Modernized aircraft which operate above 10,000 ft., usually between 14,000 and 21,000 ft.**
- **Jet aircraft which normally cruise above 30,000 ft.**

Traffic tends to congest in these three regions according to class. There is a wide range; three regions one or two altitude levels. Each one is flexible from the wind and weather standpoint. The most favorable conditions for one type is usually the most favorable for all in a particular class. The fact is that each class needs different the same altitude. The only way for a sensitive operator to operate in the favorable altitudes is to provide close spacing so that each level carries more traffic.

The same situation applies to routes. There is usually one major route desired by many flights. The end result is that the sensitivity just was available over the North Atlantic is a north. Most of the time is a compromise of a small segment of the total available airspace.

There are several different methods of doing the ocean.

- **Cost Circle**
- **Priority pattern**
- **Block fix**

- **Minimal flight path.**
- **Loose fix.**

In addition, there are various private techniques. The task of fitting these different flight paths in one map without having too many cases is extremely difficult.

Climbing Problem

The step along is another technique, a combination of best route and altitude. A departing aircraft heavily loaded with fuel will not use high power easily to transport that fuel to a higher altitude because of the high rate of fuel consumption during the climb. The flight starts low and the fuel load lightens the aircraft climb in a series of steps until the desired cruising altitude is reached.

This is a good method except that an aircraft often is too far above when out over the ocean and then finds it impossible to climb because the higher altitudes are already blocked with traffic. It is impossible to reach the destination flying at a low level so the flight comes to the appropriate point in a field or lands in route for fuel. For example, a Shearwater to New York east coast flight may be unable to climb and is forced to land at Gander for additional fuel. It is somewhat happens that the aircraft is overweight for an planned landing and cost group feel better if one land to obtain some fuel.

Flights can come from two to seven hours because of these difficulties and the need of the very often disappears. Aircraft climbing faster in the time of day. Most airlines schedule

flights at the same hour for competitive reasons. This causes 60% of all transatlantic traffic to land at Gander in less than half a day.

The combination of stringent aircraft requirements, lack of navigation aids and communication makes it quite costly to have the ocean routes as able to move as much traffic as they do. The problem for successful ocean air travel makes it mandatory to improve the present traffic control system.

Tasting New Systems

There are several devices and systems being tested in an effort to solve some of the traffic control difficulties. Forecasting results have been obtained in an experiment with ground wave propagation of radio signals. By placing transmitters near water and using soundings of the upper electronic conductivity of the ocean there has been a considerable improvement in reception.

There is also a program for examining the band MUF system to radio telephony as well as plan to install transatlantic interphones between coastlines.

At the present time there are not enough frequencies and too many air planes on one party line. Some of the airlines have radio operators who can send Morse messages, but both the International Civil Aviation Organization and International Air Transport Association require all messages to be sent by radio telephony despite the lack of facilities.



'Quiet' Sign at Newark

Newark Airport, in an effort to reduce noise and provide nearby residents, recently installed the above runway sign reminding pilots of special flight procedures designed to eliminate noise over the populated Elizabethport. The procedure set and when seen over and water, pilots have to be made with safety.



S. E. 210 Caravelle

Latest aerial photograph of the Swiss S. E. 210 Caravelle points up the close loss of Frontier's southeast coast jet program. Cracking legs of the two jet carriers is 415 mph. One was last seen in March 56 during Swiss test flight.

Frontier Asks Board Survey To Help Lower Subsidy Needs

Washington—Frontier Airlines has asked the Civil Aeronautics Board to open a general investigation of its operations to find a means of reducing the carrier's subsidy. The CAB has recognized the need for strengthening of smaller carriers. The carrier believes that the route awards in those cases were made to subsidize trunk airlines, with the exception of Continental Air Lines.

Frontier points to recent decisions as the New York Chicago case, the Denver service case and the Southeast Northwest service case as indications that CAB has recognized the need for strengthening of smaller carriers. The carrier believes that the route awards in those cases were made to subsidize trunk airlines, with the exception of Continental Air Lines. Frontier feels that the Civil Aeronautics Act applies equally to local service carriers and trunk carriers, and that CAB should use the same rules to help local airlines get off subsidy that were applied to the trunks.

Seek Improvements

The carrier suggests the Board cover seven questions in its investigation, including the matter of whether non-trunk routes should be closed and whether trunk carriers should be suspended at certain Frontier points. Frontier also wants the Board to decide whether air route structure should be adjusted as expanded.

In the proposed investigation, Frontier would like CAB to discontinue what showing should be made for a trunk carrier route to extend its operations in the best interests of Continental Air Lines, Eastern Air Lines, Trans World Airlines, American Airlines, Western Air Lines, Braniff Airways, Bonanza Airlines and Northwest Airlines which blanket its area of operation and could cripple Frontier's opportunity for normal development.

Frontier also wants CAB to look into other means of stopping its system, such as operating airways "which will occasionally permit the use of non-allocated space." Frontier's route system present route problems not faced by other local service carriers. Frontier shows the lowest population per stop and the highest ratio of passengers bonded per 1,000 population.

While the system passenger load factor was 46.19% for the last nine months of 1955, the carrier estimates it would average a load factor of more than 35% to break even under present operating conditions.

Special Problems

Frontier operates over a route area which stretches from the Mexico to the Cowboy borders and includes Montana, North Dakota, Wyoming, Utah, Colorado, Arizona and New Mexico.

Most of the routes in the area is rugged and makes regular transportation difficult. This means that in a number of instances Frontier often the only transportation available between the points it serves.

While the character of the area served contributes to a substantial traffic loss and cost it also means a light distribution of population and few large traffic markets to support work schedules.

Utah presents special operating problems for Frontier's fleet of 12 DC-3's. Sharp altitude and temperature difficulties mean weight and payload penalties. On some flights, Frontier has times such low altitude passes at Therman and Terman to high altitude points in Colorado.

A lot of the mountainous airports served are above 7,500 ft., and some flights must be operated at high as 15,000 ft. to clear obstructions.

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Nonsked Buys 1049-He

United States Overseas Airlines has ordered two Lockheed 1049 He Super Constellation convertible passenger-freight transports at a cost of \$4 million. Initial delivery is scheduled for May 1957, although one of the planes may be delivered at an earlier date, a company official stated.

Chief Aircraft Board's recent award for awarding scheduled services (AW No. 2), p. 12) was an important factor in the airlines' decision to buy 1049-He's, he indicated. U. S. Overseas Airlines currently operates one DC-6 from Cape Mue (N. J.) County Airport. He said that it expects to leap the DC-6's after it gets the new 1049 He.

CAB Orders

GRANTED

State of Ohio leave to reference at the North Central Lake Central negotiation case. **United Airlines** in agreement to carry out its plan from New York to Indianapolis, and Jan. 1, 1956.

Carolina Airlines, Chicago versus a long as other agreed to operate between Chicago and Miami to Lake Frontiera City and Havana.

APPROVED

Agreement between various carriers accepted by the International Air Transport Assn. officials to Mont Adria (split) conference, May.

Reconciliation between various carriers accepted by the International Air Transport Assn. relating to rate and traffic matter.

ORDERED

Supplies and investigation of a reduced rate between Chicago and Tampa filed by Peninsula Air Transport. **Lake Central Airlines** to show cause why it and one for its parent beginning Dec. 2, 1955 should not be set at a rate of 27.5 cents per enroute mile.

Supplies and investigation of reduced rates between Dallas and Oakland and between Miami and Newark. **New York, Fla.** depts and Washington filed by Aero Flight Carrier Corp.

DEFERRED

Plung Tiger Line's application for an exemption of 10 percent 2-month charter flight between New York and Moscow, since the ITL application does not conform to the spot or requirements of the Trans Atlantic Charter Policy.

Reopened cases in Pacific-Pacific case, since the airport matter in question has been dropped through the issuance of General Air Bus. permanent schedule.

DEFERRED

Alleged Airlines' application to succeed award of Baltimore, Md. between Nov. 1, 1955 and Apr. 1, 1956.

Aero Air Cargo Inc.'s petition to expand the international air freight forwarder as transportation to include the possibility of a

goal of estimate of public convenience and necessity. Petition for leave to intervene in the case are presented for American Airlines, Eastern Air Lines, Delta Air Lines, Pacific Northwest Airlines, Pan American World Airways, Republic Airlines, Seaboard and Western Airlines, Trans World Airlines, American East Airlines, the Aircraft Transport Assn. and the Air Flight Forwarder Assn.

Lake Central Airlines' petition for acceptance of a Bond division during Lake Central an acceptance to operate between Cincinnati and Zanesville, Ohio.

Shortlines

► **British European Airways** had a net profit of \$6,880,000 in the period April 1 to Sept. 10, 1955. In the six-month period, BEA flew 453,100,000 passenger miles, 27% more than the same period of 1954. Load factors were 72.7% in international operations and 78.5% in domestic service, compared with 70.2% and 74.8% for the 1954 period.

► **31 Al Burt Airlines** has purchased two additional Constellation, one of which is to replace the plane shot down in the Bulgarian Sea.

► **Flying Tiger Line** flew 6,523,888 ton-miles of air freight in November, and \$1,826,501 ton-miles of freight in the first 11 months of 1955.

► **Lufthansa German Airlines** has appointed a sub-office in Detroit. Other offices are operating in New York, Chicago and San Francisco.

► **North Central Airlines** had a net profit of \$1,331,000 in November, bringing the net for the first 11 months to \$15,761,000 after taxes. Net profit for the 1954 period was \$13,736.

► **Subera Beluga World Airlines** plans a \$250,000 expansion program which will give the airline close transportation in 25 cities in Canada, the United States and Mexico. The expansion is to include facilities in Toronto, Mexico City, Denver, Boston, Pittsburgh, St. Louis, Kansas City, Hartford, Houston, Memphis, Tulsa, Minneapolis and Havana, Cuba.

► **Seaboard and Western Airlines** flew 1,512,888 ton-miles of international freight in November, an increase of 8% over November, 1954.

► **Trans-Canada Air Lines** reports passenger traffic grew 13% in 1955 while capacity was increased 16%. TCA flew 51.4 million passenger-miles in domestic service and 151 million passenger-miles in international operations last year.

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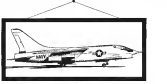
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Approach Visibility

AMERICAN WIRE's continued interest in the application of "radar to the runway" is being better understood and increasingly recognized by business. On May 13, 1951, AW announced a new system that charts an accurate picture of visibility at the approach, regardless of tower location. Just recently, on the December 1, 1951 issue, AMERICAN WIRE published C-45's report of the Beech Aeronautics accident that occurred last Feb. 4. C-45's 360 whole coverage wing Chief, William A. Hugg, stated an instrument approach system, such as the one being developed, would have prevented such a tragedy.

C-45's report concluded: "A probable cause of the accident was instrument divergence caused by the loss of visual reference during the final visual phase of the approach. Visibility was measured as of descent at an altitude less low to effect descent."

Exactly as the report, C-45 states: "It is not likely the accident was prevented if a low level was used. It is believed that this was evidenced by the pilot's report of descent at a position to clear them."

"The U. S. Warbler Bureau has obtained many sets of end-of-the-runway instrument readings, government and a transmittance reading, instrument, illumination of the runway, instrument, illumination of the runway being considered."

A primary test with high volume traffic airports showing test procedures. As a result, instrument test should have included and is in operation at the Chicago Midway Airport. The transmittance has been installed but has not yet been installed as of October 1951.

"The program for the installation of the system of these instruments is under development and will continue during the next year. The final test will be made of the system in the next few months. The Board wishes to endorse the progress and transmittance that it proposes as a possible solution."

Developed by the Bureau of Standards and built and maintained by Consolidated Companies. The transmittance consists of a primary receiver of radio waves, which is installed along the runway, the projector sends a concentrated light beam along a 500-foot run to the receiver. Wave light beam is picked up by a lens system in the receiver and launched into a photoelectric element located in the receiver lens. The lens emits an electronic signal that a amplified and transmitted from the aircraft

to the instrument receiver in the control tower where actual runway visibility is checked continuously.

Even though visibility conditions are not at the runway, the instrument receiver conditions at the control tower, the instrument receiver has a pilot approach by a landing. When the light on the tower is on, as the radio wave beam reaches receiver and still waves continue readings from the system. Designed to work in conjunction with the transmittance, the existing beam receiver—also located in the approach receiver—receives the "radar" light rays, and, through the photoelectric element, relays the approach data to the instrument receiver. The instrument receiver relays the information to the pilot's cockpit.

J. S. HARRIS
Manager, Illumination
Consolidated Companies

'No Go' Refunds?

Each week I enjoy your publication—keeping up with the business. I have just completed a course, planned by myself, in the country, visiting aircraft manufacturers. It is a credit to an article that it was able to keep me entertained, even though much of my time was spent visiting manufacturers instead of sleeping in hotel rooms—which I had reserved, paying extra.

From time to time I read reports about the airlines concerned over the "no show" and the "no show" BUT as the airlines are going on, things to the "no go" problem. Can this be done, please, as well as "no show" etc. and more progress how? Was this one depend on one "no show" etc. and more progress how? Was this one depend on one "no show" etc. and more progress how?

M. V. ROSENBERG
2211 Venable Road
Vestal, New York

Airport Advertising

Thanks to the initiative of Federal, state and local governments we are fortunate to have numerous excellent new airports and terminal facilities. Except those at the Philadelphia International Airport and the Washington, Mass. airport which I avoid myself. Both are beautiful in every detail and excellently built in every respect. What makes me however is the picture of a new Chicago airport automobile designed in the lobby. I ask: Is it so the best interest of aviation to use airports and buildings financed with the money for such commercial purposes? It questions the propriety of such public property in further public interests.

Bob STINEBAUGH
130 Cabell Avenue
Cullmanville, N. J.

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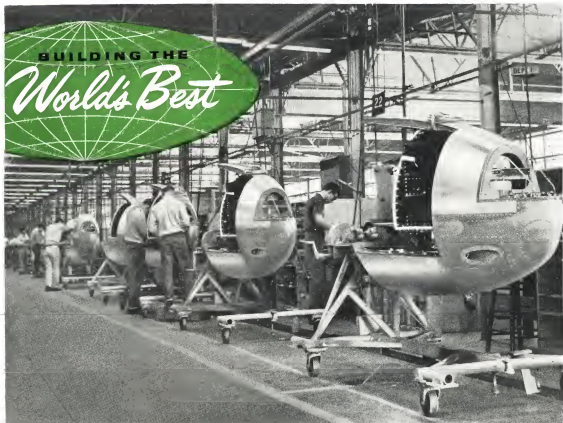
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